



Design and Development of Radiation Hardened Application-Specific Integrated Circuit for the ATHENA Mission

APC PhD Day - November 15, 2017

Si CHEN

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Next Generation X-Ray Space Observatory of ESA

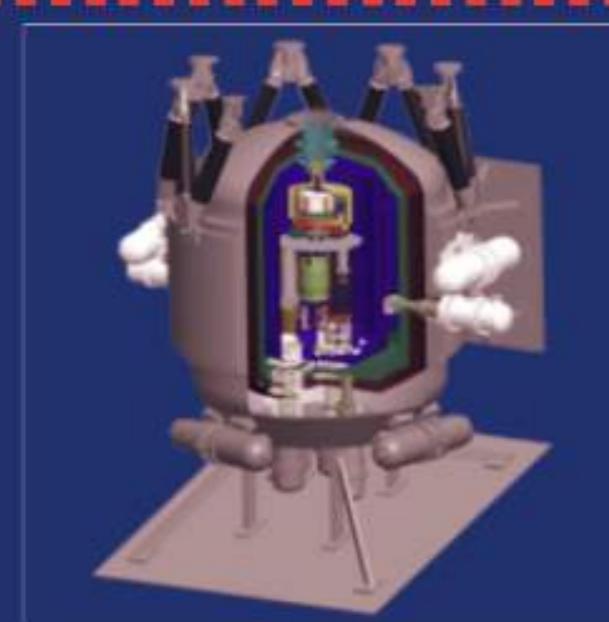
Ariane V class launcher

Satellite mass ~ 5500 kg

Power ~5600 W

Focal length: 12 m

Lifetime: 5 years (10 years)



X-ray Integral Field Unit: X-IFU

ΔE : 2.5 eV

Field of view: 5 arcmin

Large array of TES cooled at 50 mK

Barret et al. 2013 arXiv:1308.6784

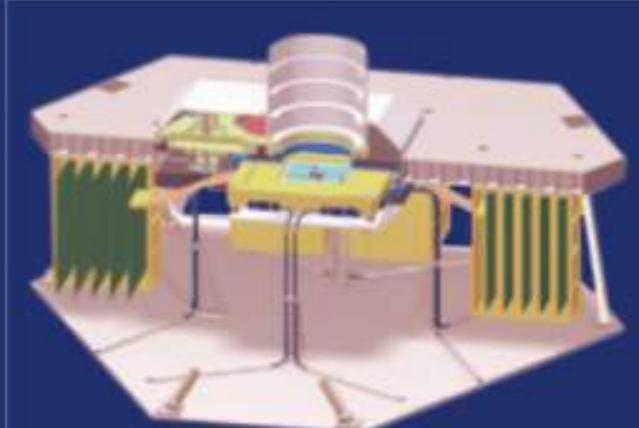


Silicon Pore Optics:

Effective area: 2m^2 @ 1 keV

PSF (HEW): 5''

Willingale et al. 2013 arXiv:1308.6785



Wide Field Imager: WFI

ΔE : 125 eV

Field of view: $40' \times 40'$

Rau et al. 2013 arXiv:1307.1709



ATHENA Mission

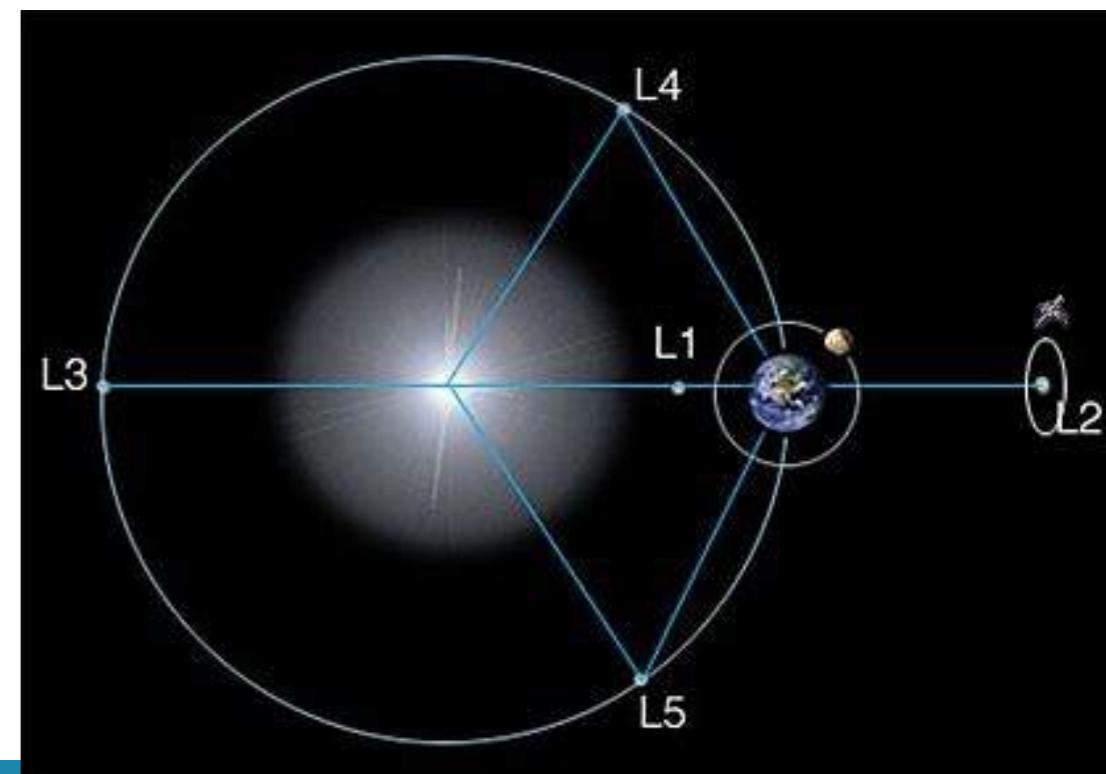


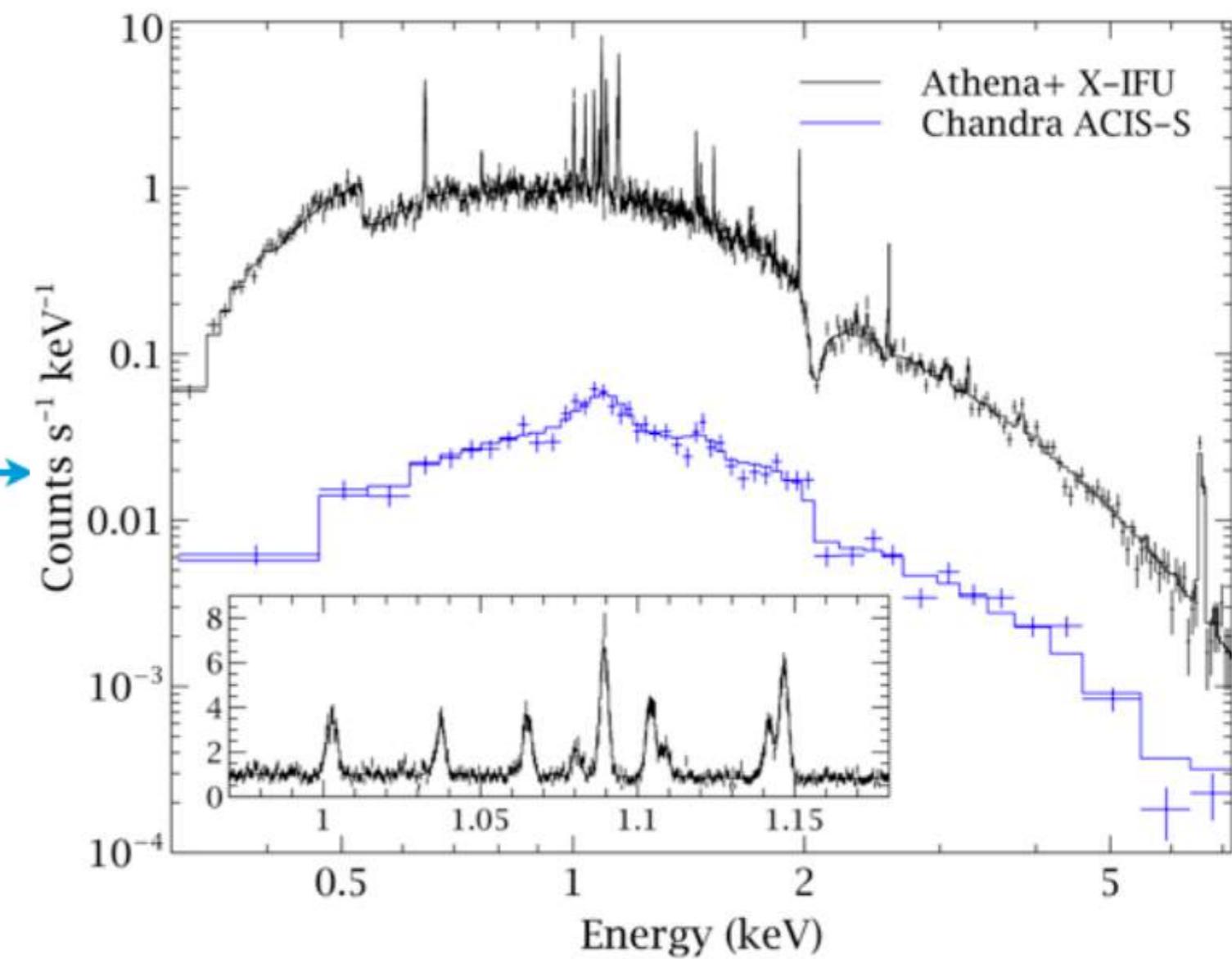
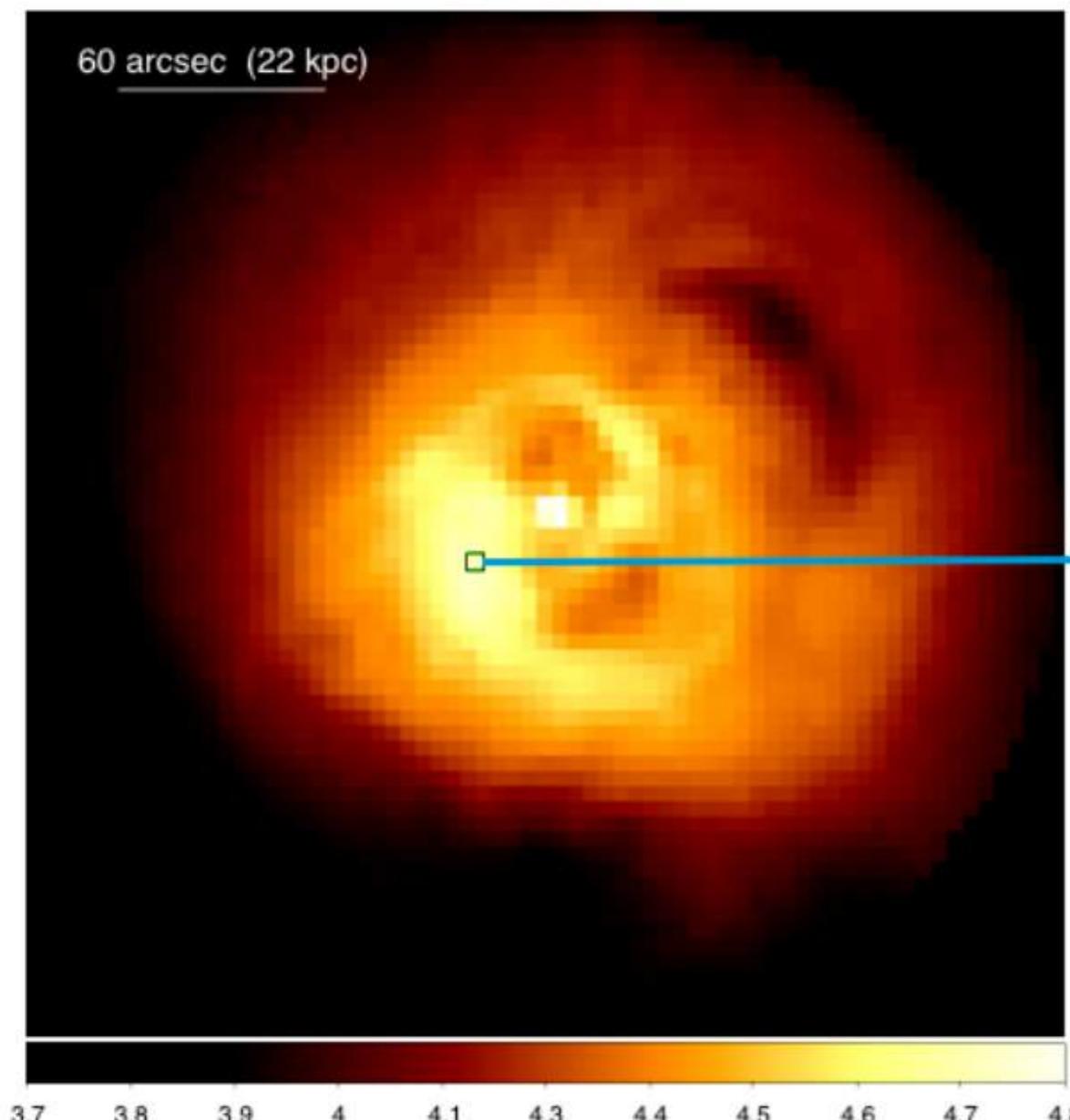
- ❖ « Hot and Energetic Universe science » theme
 - ❖ How does ordinary matter assemble into the large-scale structures we see in the Universe today?
 - ❖ How do black holes grow and shape the Universe?

- ❖ 0.5-12keV range
- ❖ L-class mission , 5 years lifetime, Launch date: 2028-2030
- ❖ Halo orbit around Lagrange point L2,



ATHENA+ COLLABORATION

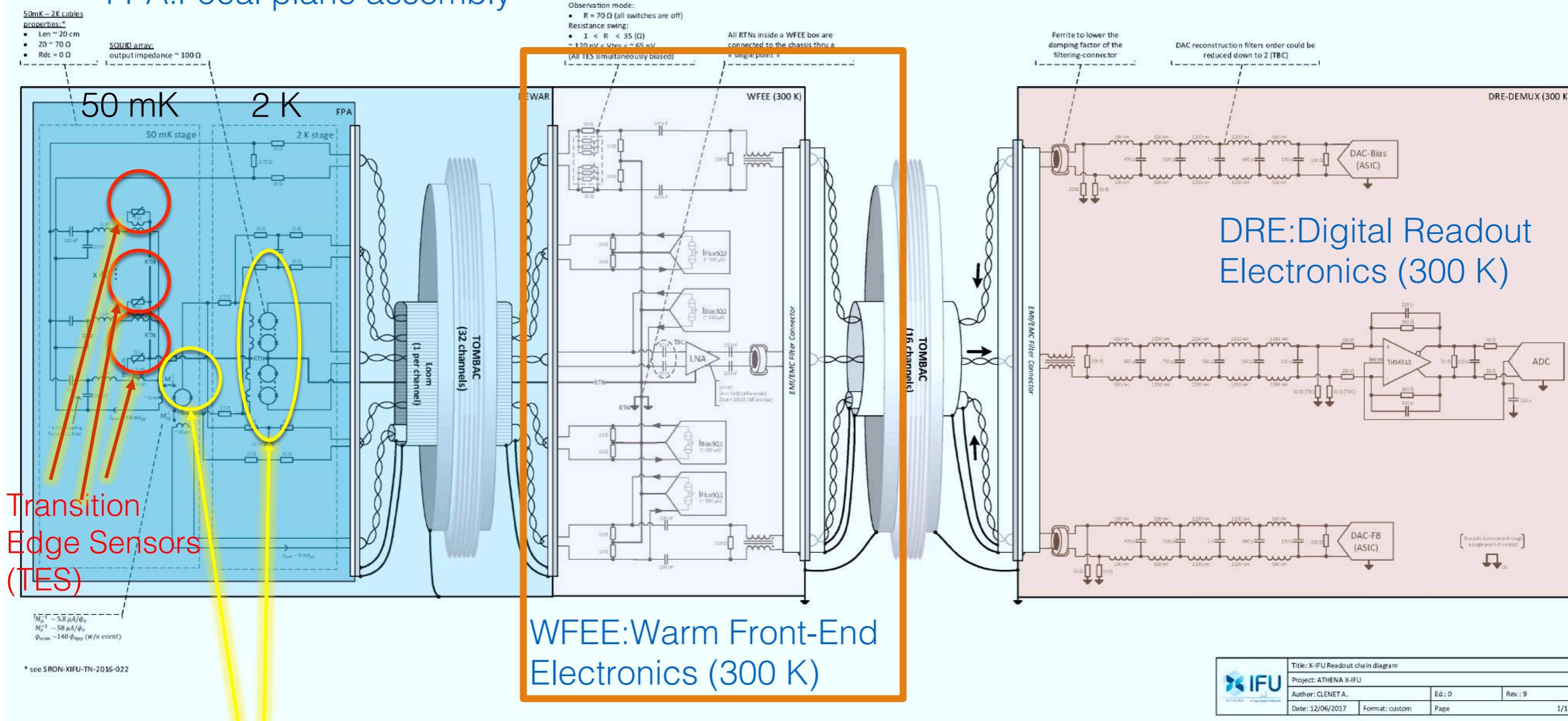


 Simulation of a 50 ks X-IFU exposure of the Perseus cluster

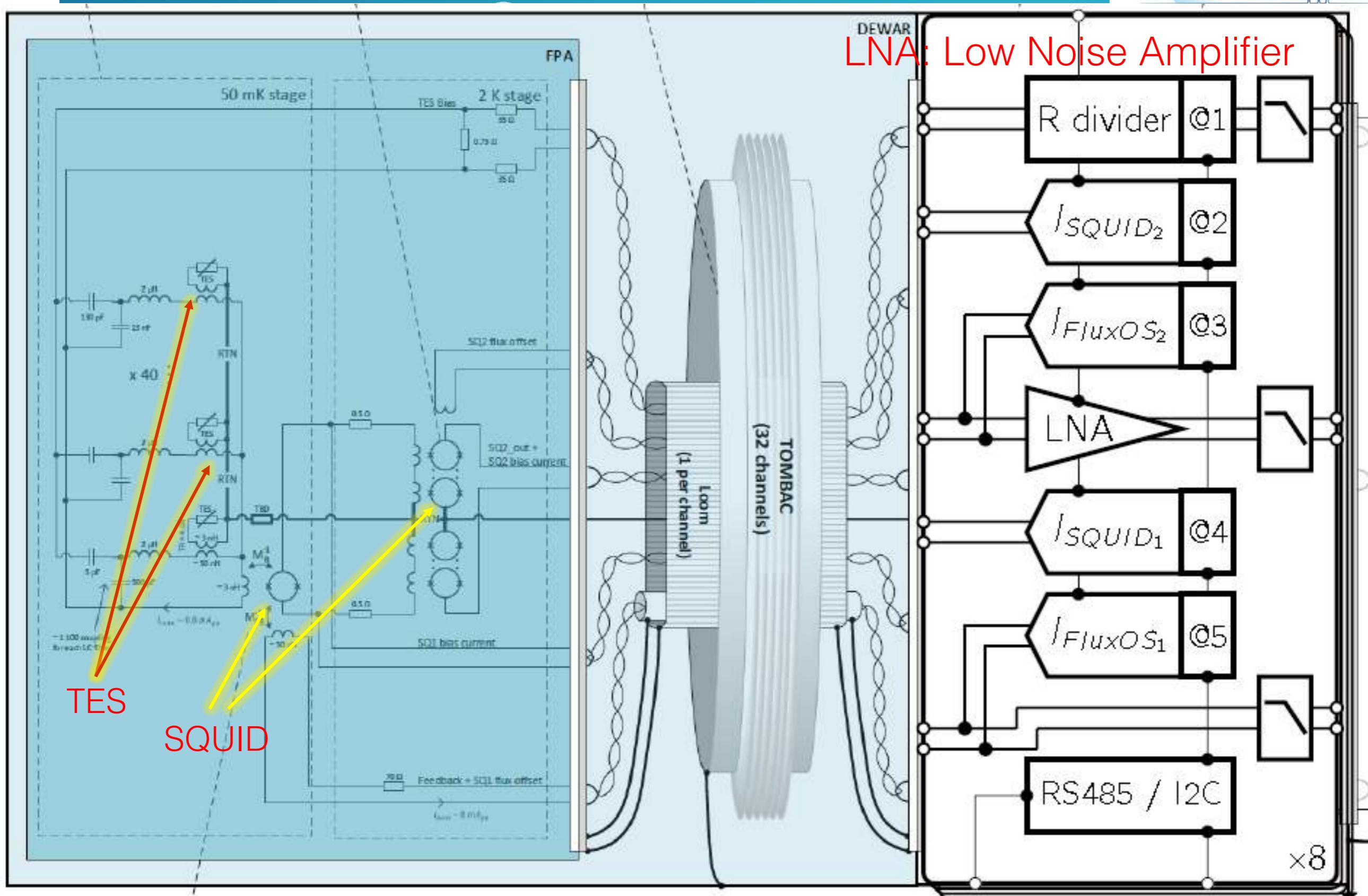
Croston, Sanders et al. (2013, Athena+ Supporting Paper)

- ❖ Frequency-Division Multiplexing(FDM) 1-6 MHz, factor 40
 - ❖ Matrix of $3840 \text{ TES} / 40 = 96$ readout channels

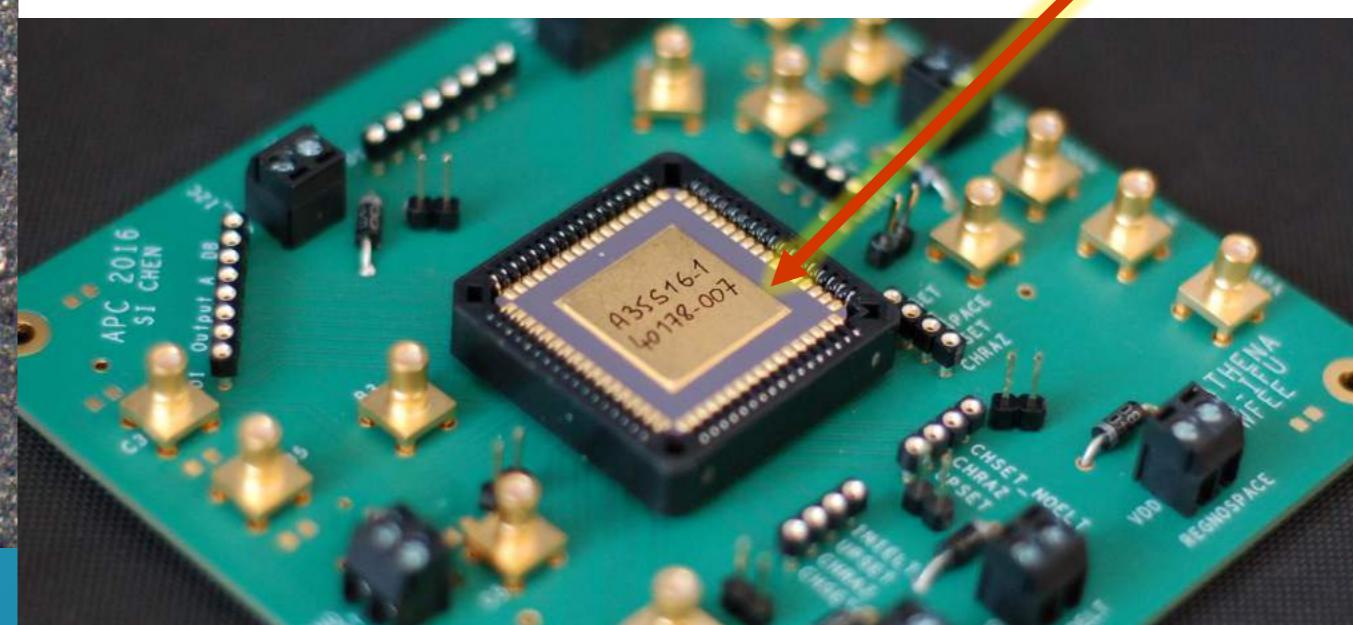
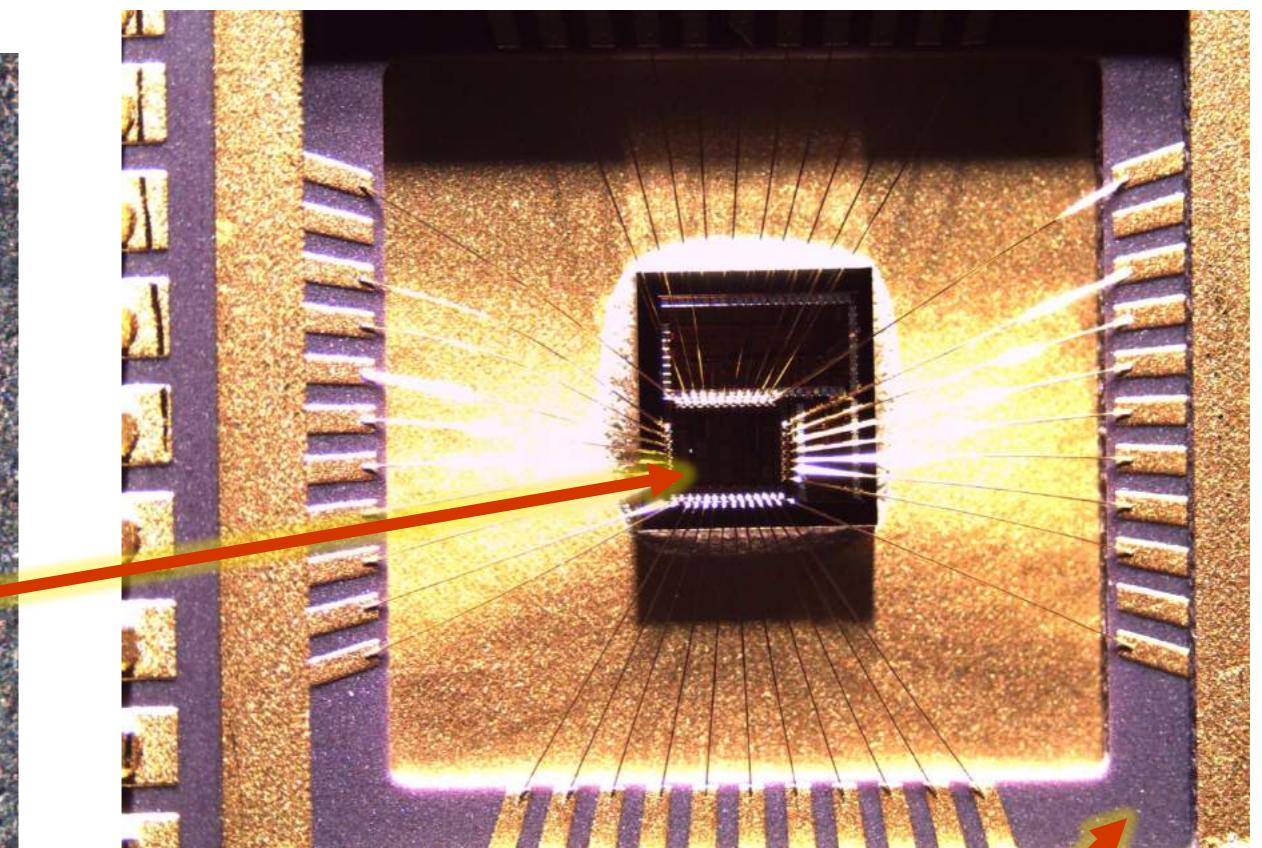
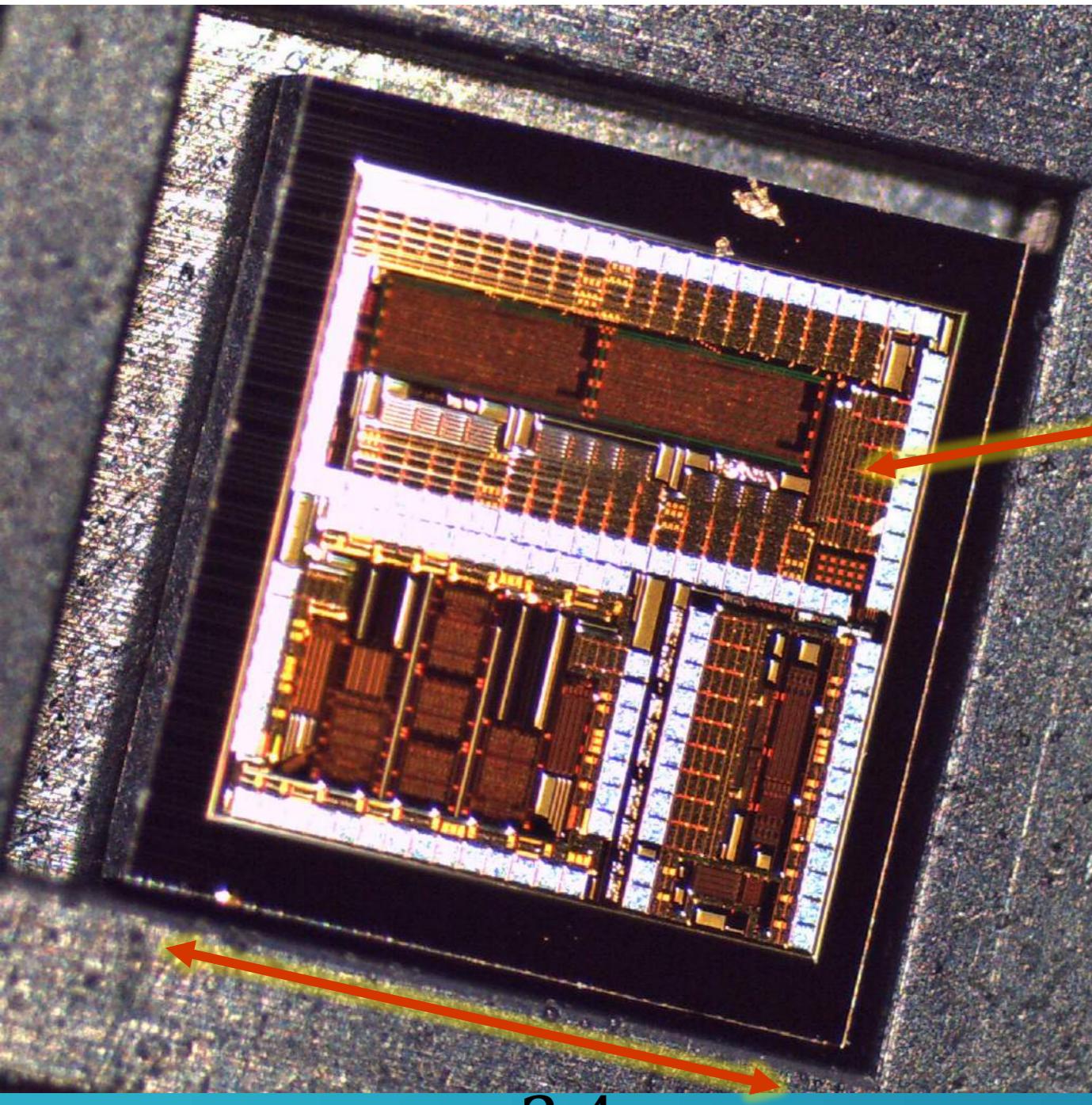
FPA:Focal plane assembly



SQUID: Superconducting QUantum Interference Device



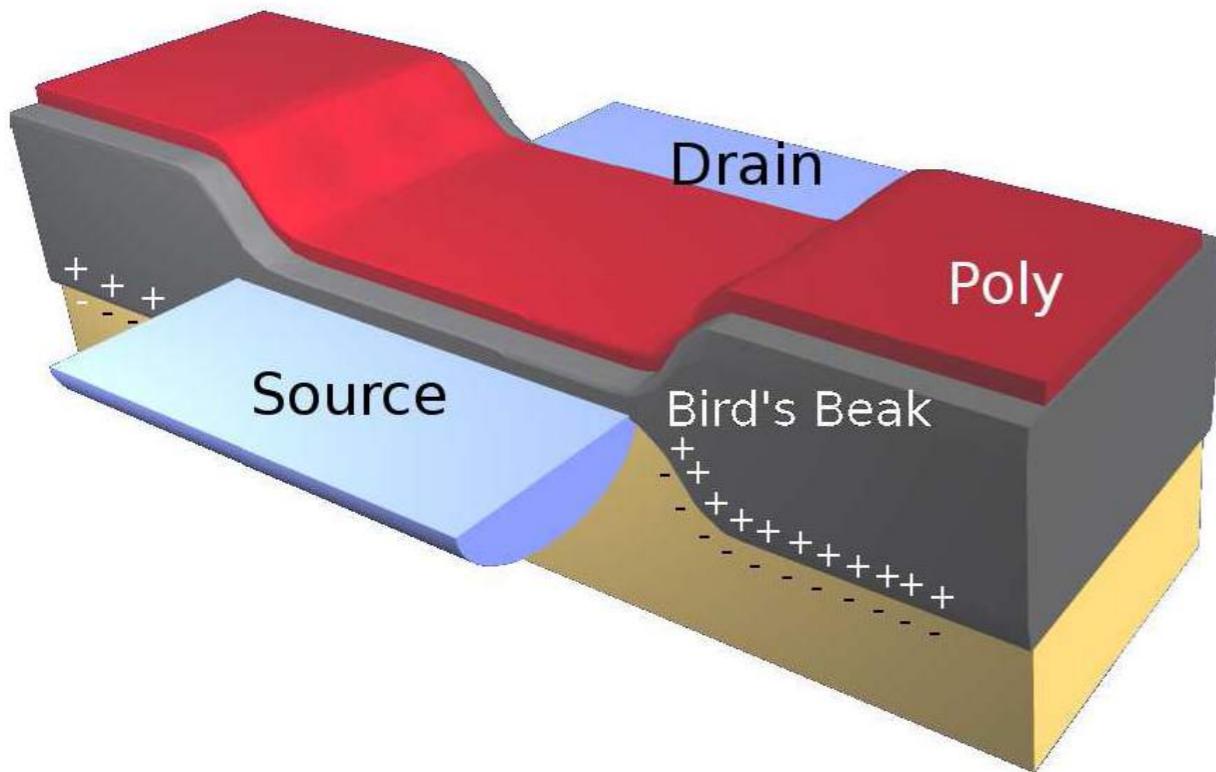
- AMS 350nm BiCMOS SiGe technology: BJT+MOS, good TID (Total Ionizing Dose) tolerance, widely used in space instrumentation, reliable
- ASIC: Application-specific integrated circuit



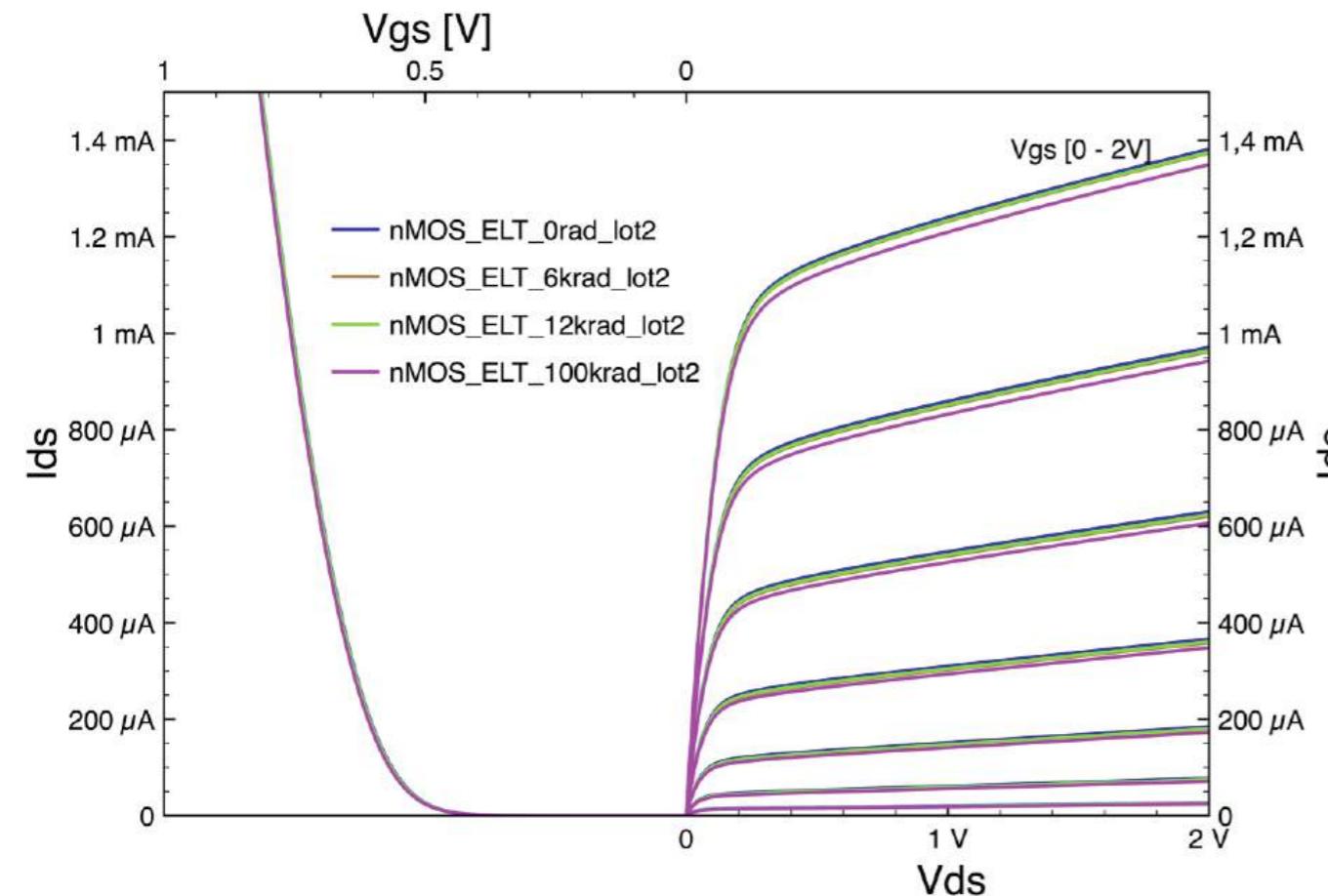
- ❖ TID(Total Ionizing Dose): **accumulative** effect
- ❖ γ source Cobalt 60 with 1.25 MeV, COCASE, CEA-Saclay, >1 months
- ❖ 1.4-7 krad during the 5 years lifetime of ATHENA at L2 point

Pack 1	0 rad	1.7 krad	6 krad	8 krad	10 krad
Pack 2	0 krad	6 krad	12 krad	100 krad	

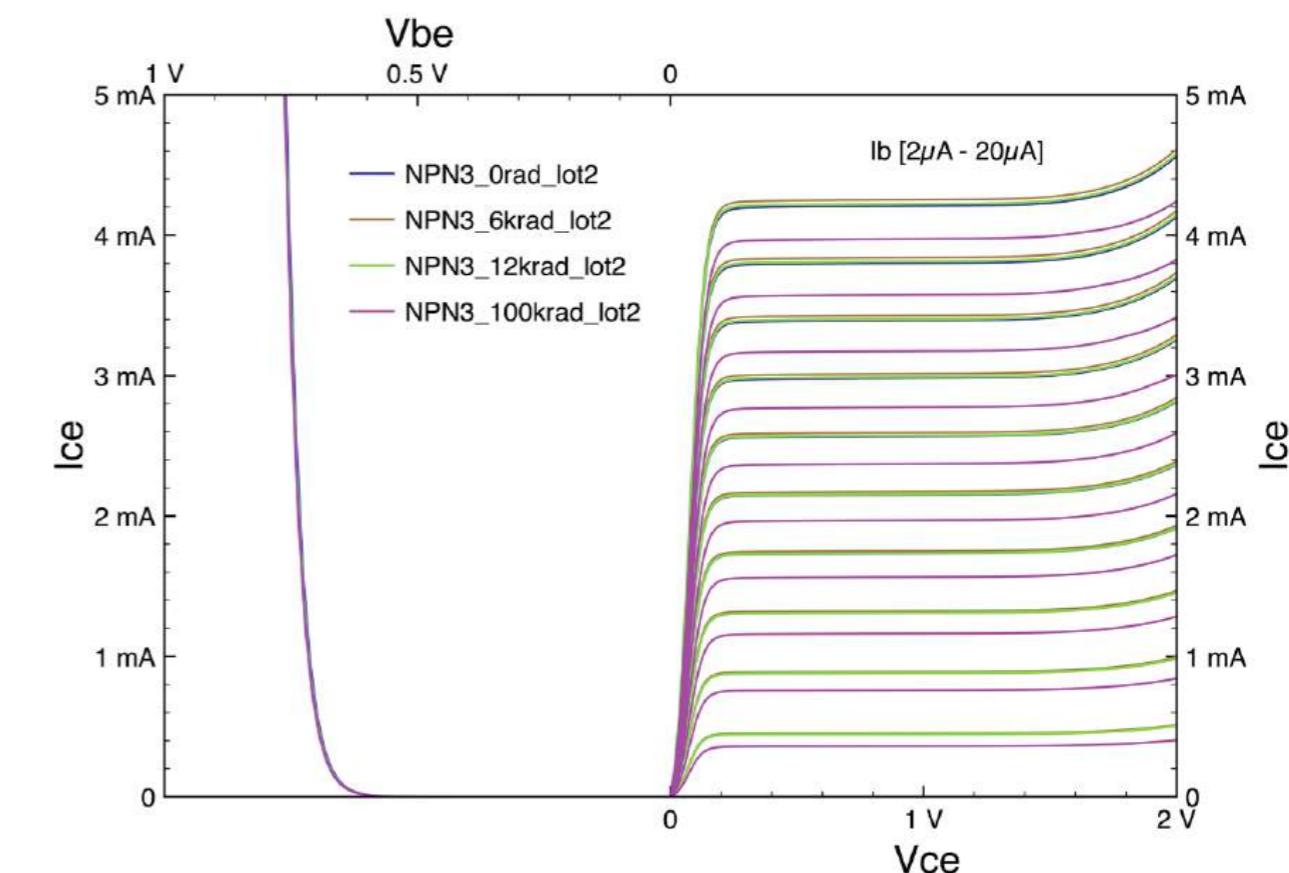
- ❖ Positive charged particles trapped in oxide may cause a noticeable leakage current between drain and source, thus change the DC characteristics of MOS



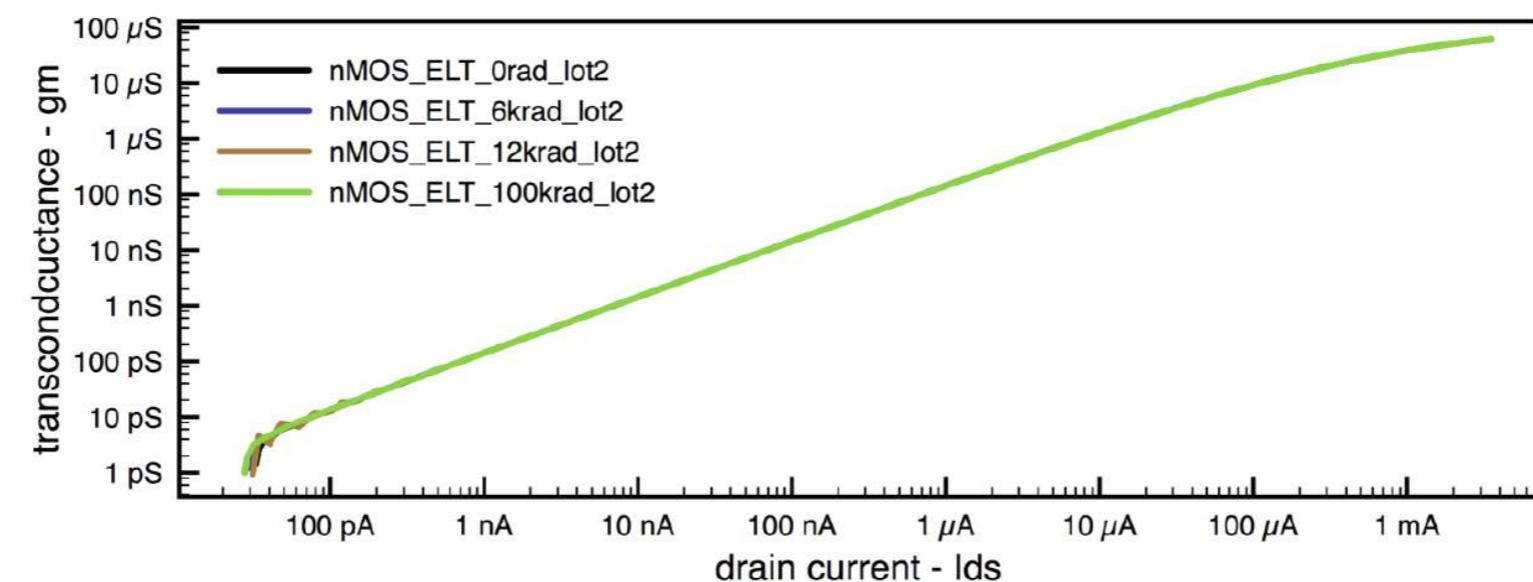
NMOS ELT $I_{ds}(V_{gs})$ & $I_{ds}(V_{ds})$



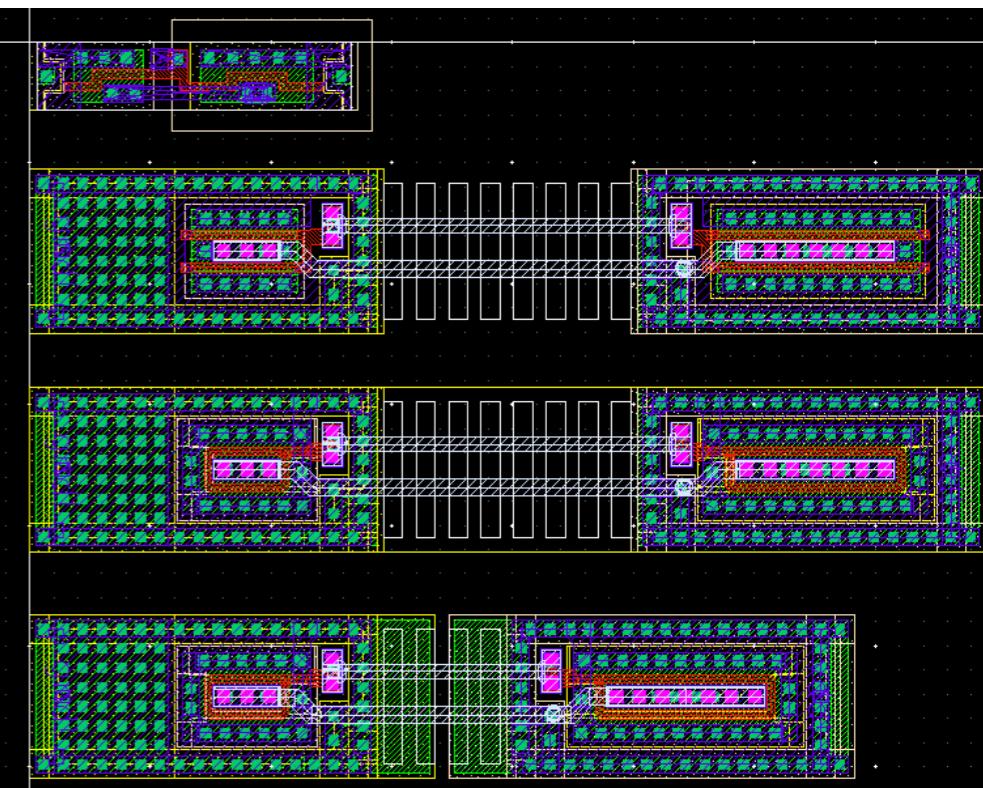
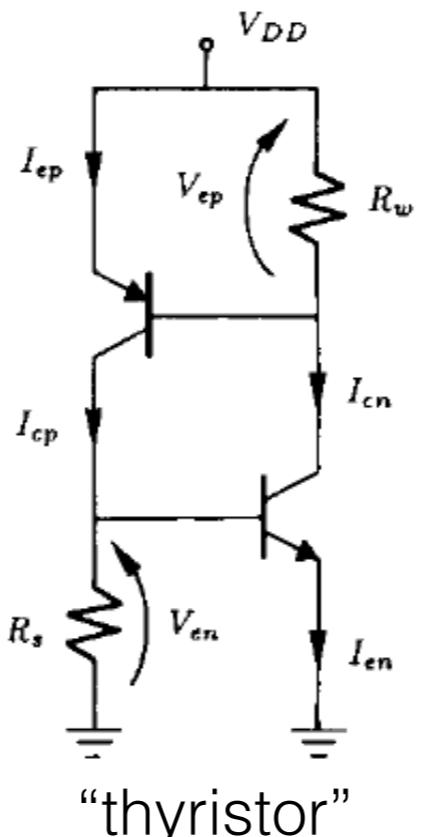
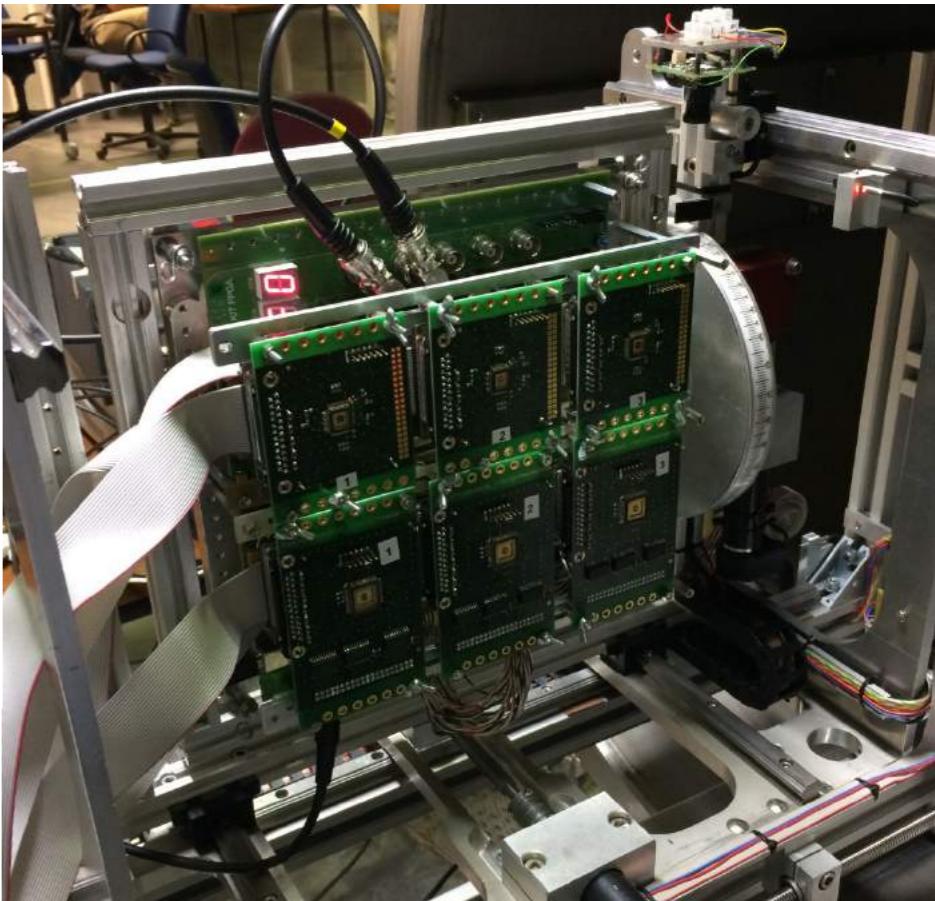
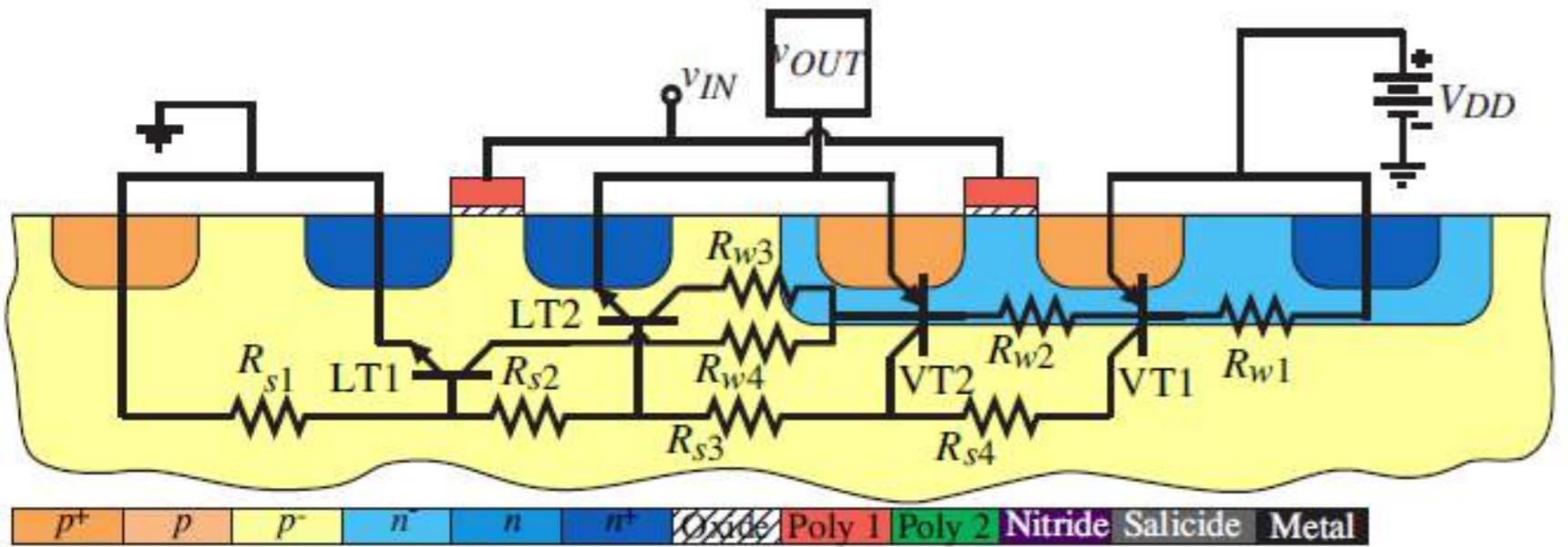
npn BJT $I_{ce}(V_{be})$ & $I_{ce}(V_{ce})$



NMOS ELT transconductance $gm = dI_{ds}/dV_{gs}$



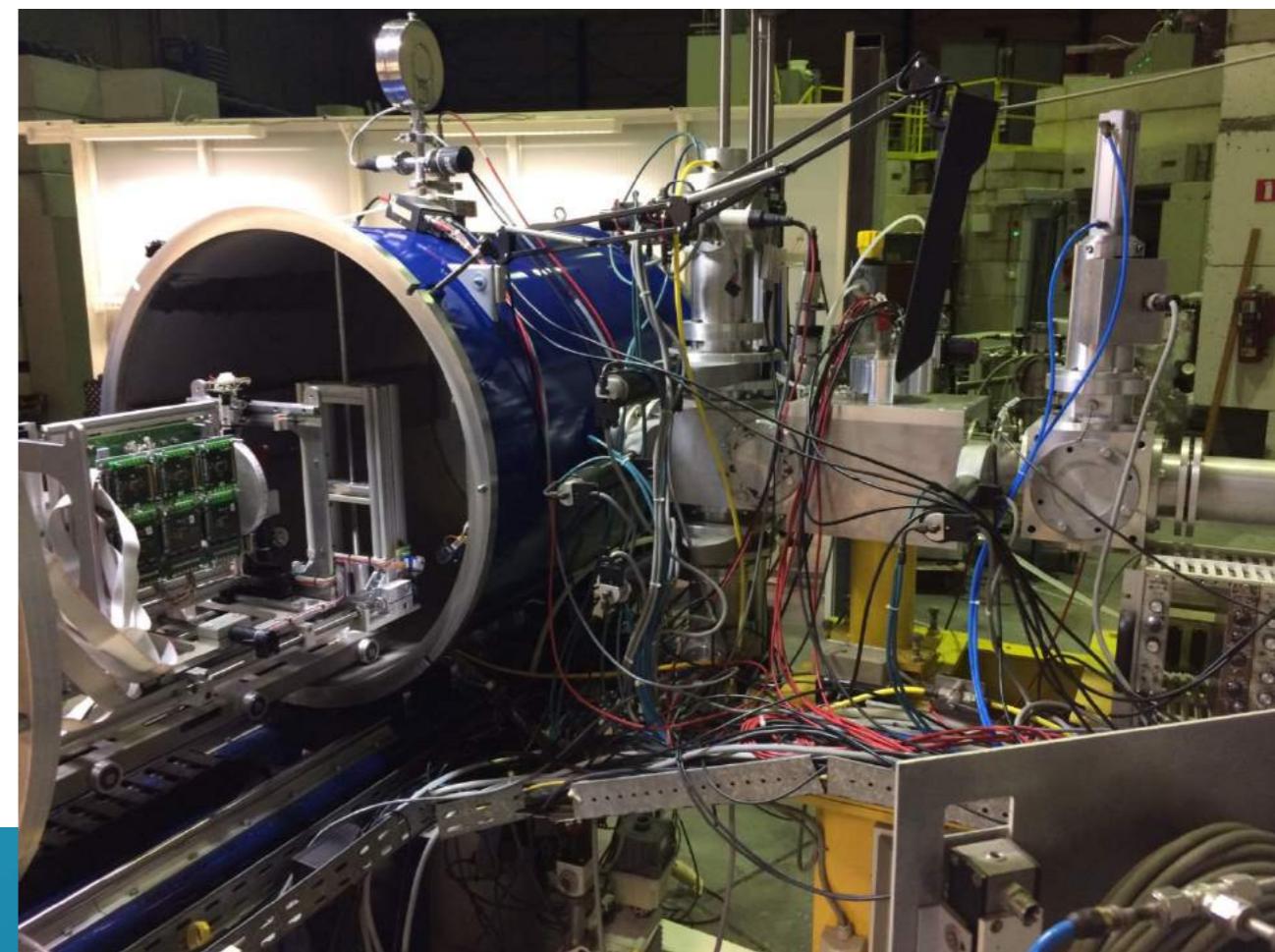
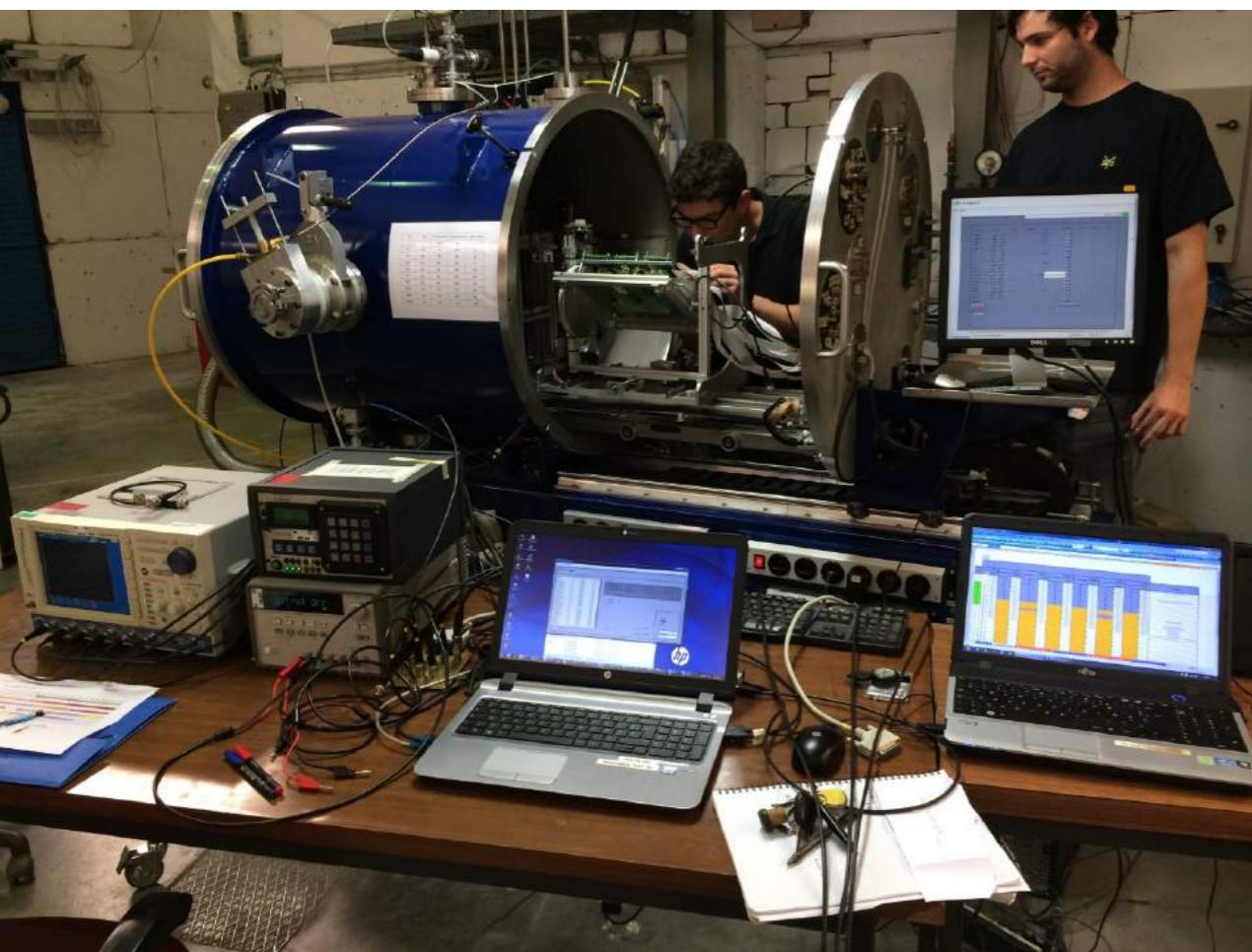
- SEL(Single Event Latch-up): caused by heavy ions or protons from cosmic rays or solar flares

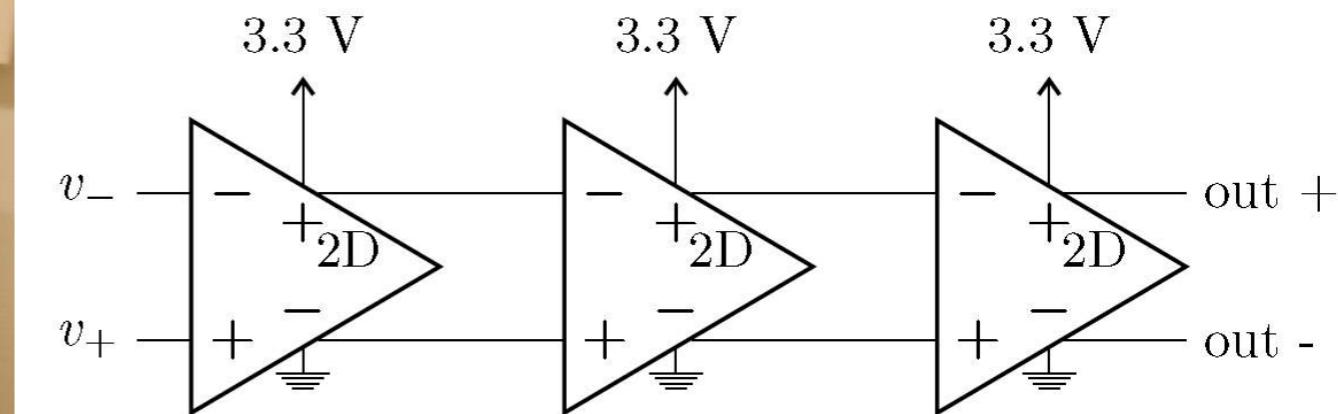
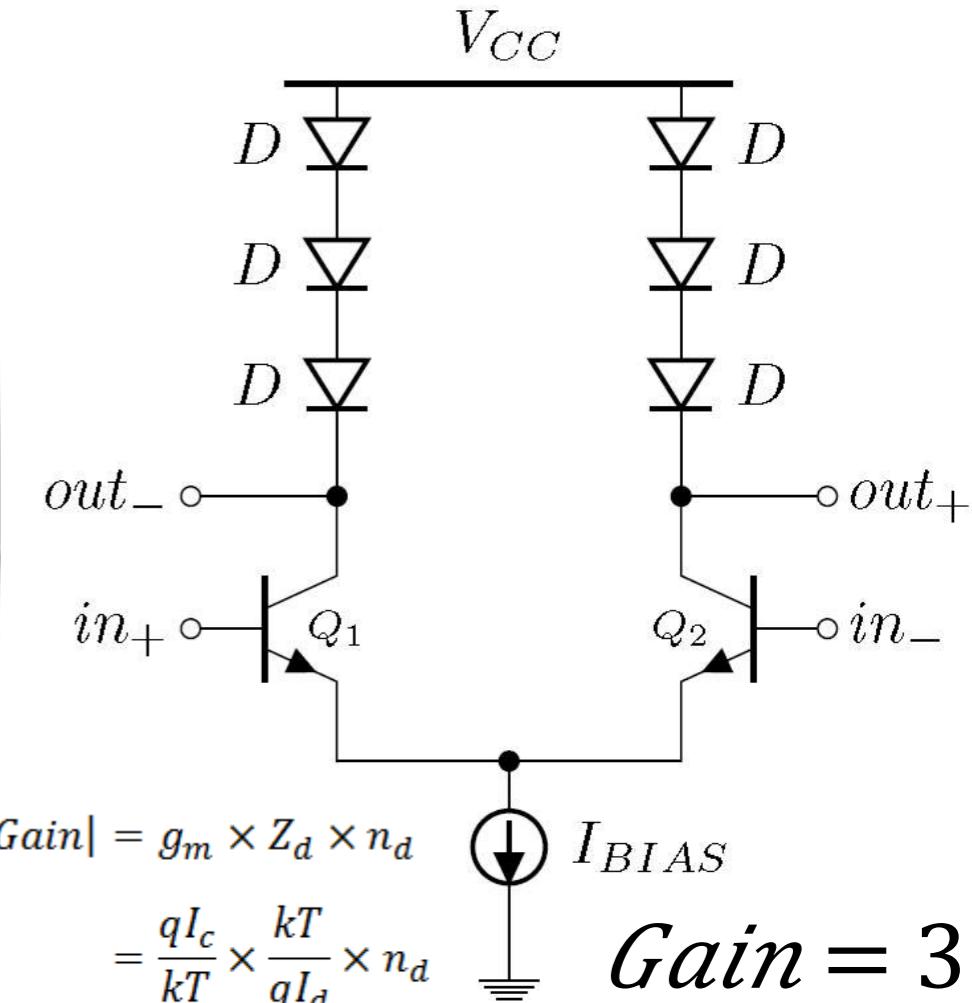
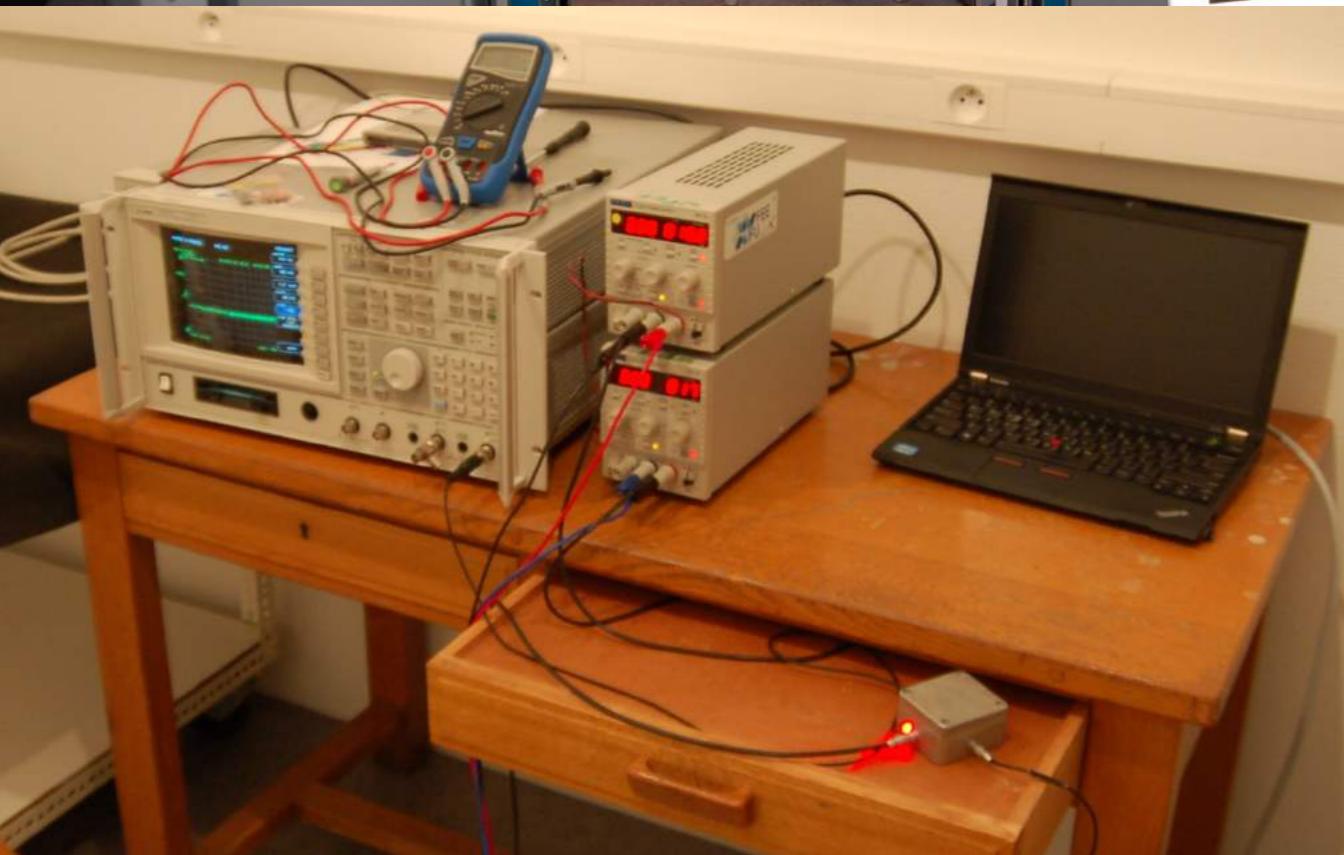


- "Cause I'm the one who's gonna make you **burn**"

Cyclotron, Universite Catholique de Louvain, Belgium

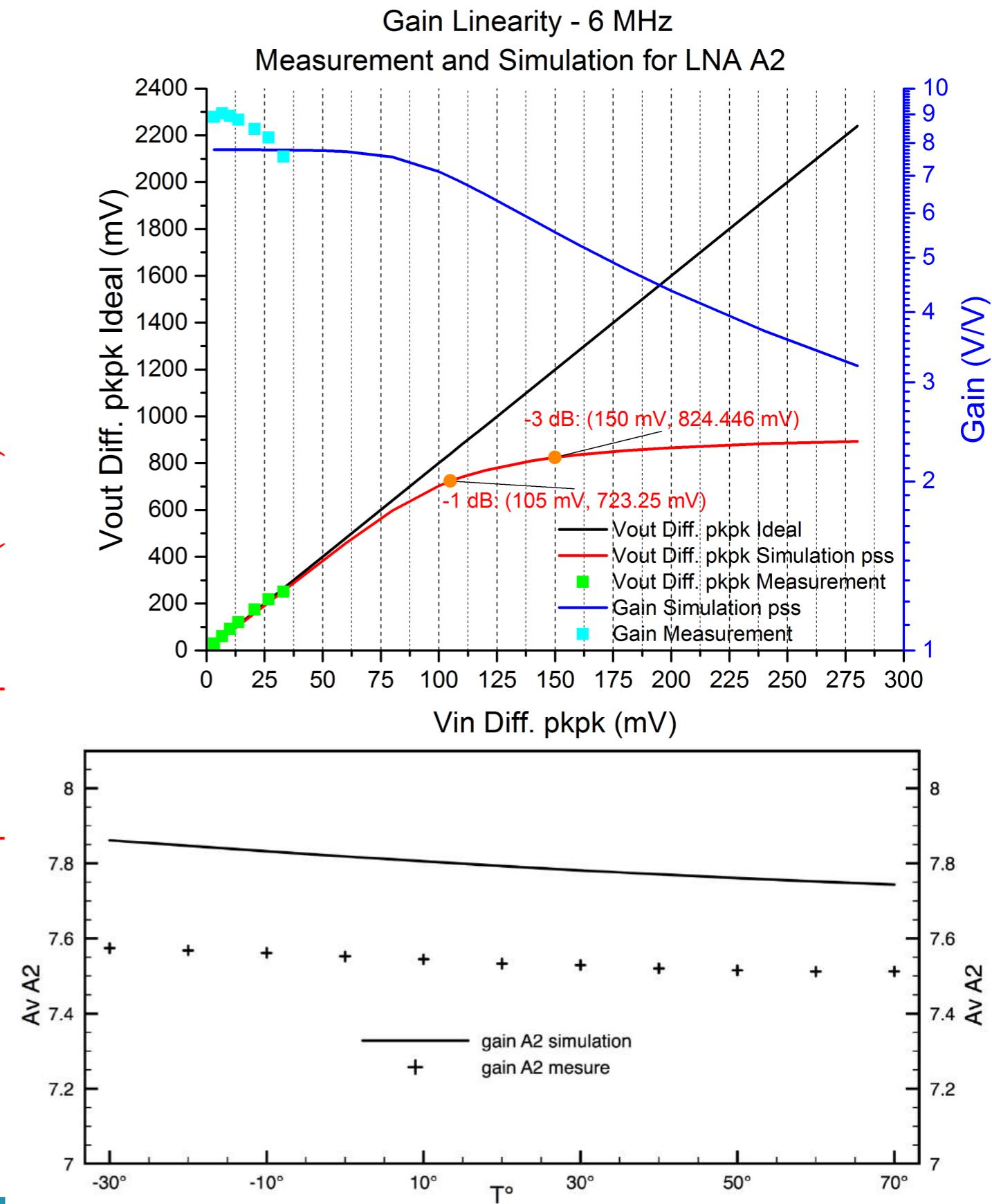
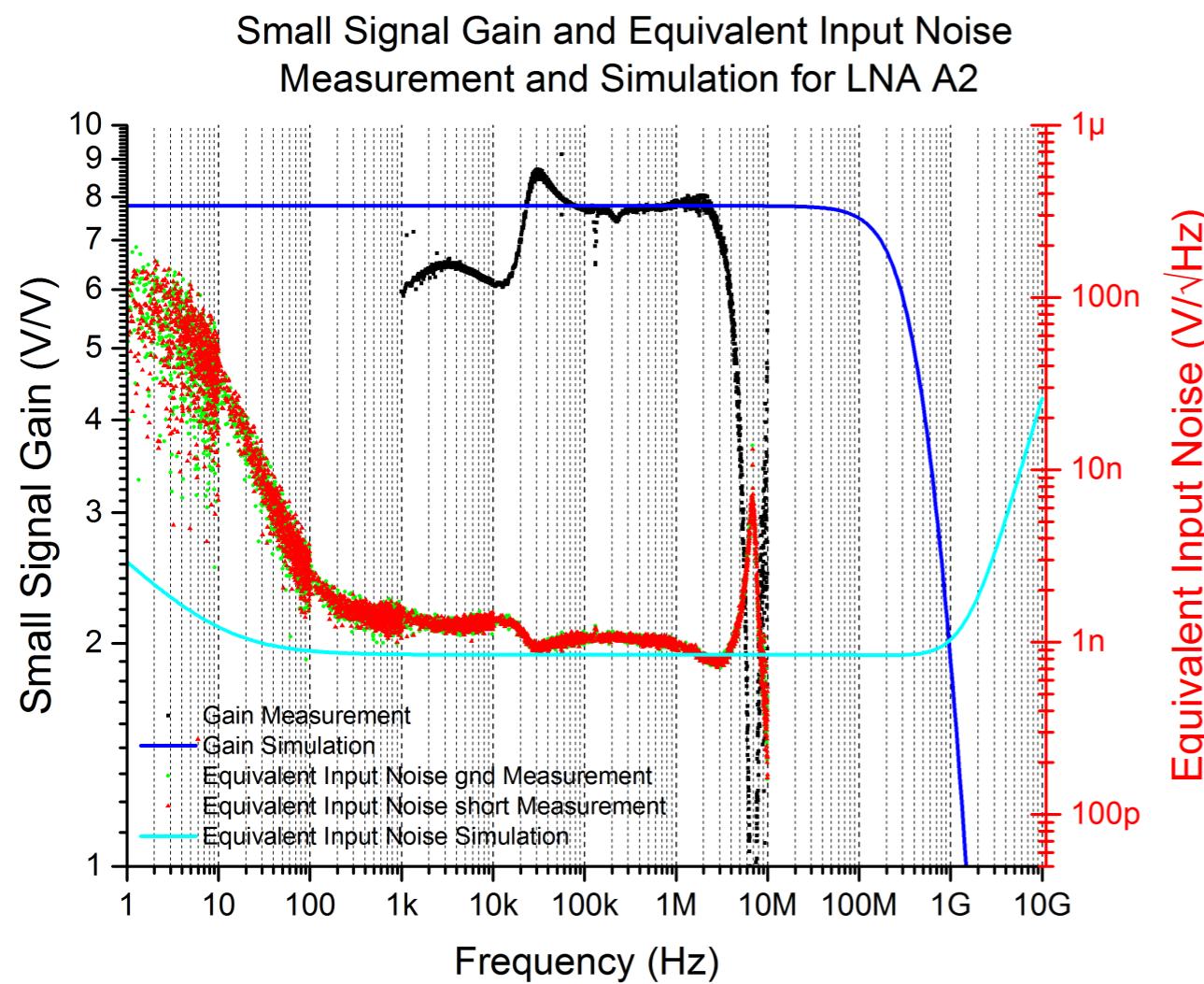
Ion	M/Q	DUT energy [Mev]	Range [$\mu\text{m Si}$]	LET [MeV/mg/cm²]
Carbon $^{13}\text{C}^{4+}$	3.25	131	269.3	1.3
Neon $^{22}\text{Ne}^{7+}$	3.14	238	202.0	3.3
Aluminium $^{27}\text{Al}^{8+}$	3.37	250	131.2	5.7
Argon $^{40}\text{Ar}^{12+}$	3.33	379	120.5	10.0
Chromium $^{53}\text{Cr}^{16+}$	3.31	513	107.6	16.0
Nickel $^{58}\text{Ni}^{18+}$	3.218	582	100.5	20.4
Krypton $^{84}\text{Kr}^{25+}$	3.35	769	94.2	32.4
Rhodium $^{103}\text{Rh}^{31+}$	3.32	972	88.7	45.8
Xenon $^{124}\text{Xe}^{35+}$	3.54	995	73.1	62.5





$$\text{Gain} = 2^*2^*2 = 8$$

- Gain ≈ 8
- Noise $\approx 1 \text{ nV}/\sqrt{\text{Hz}}$





WFEE identified Specifications

	Identified spec	ASICv1 measure	ASICv2 goal
Gain	80 V/V	3, 8, 24 V/V	
Noise	1 nV/ $\sqrt{\text{Hz}}$	\approx 1 nV/ $\sqrt{\text{Hz}}$	
BW	1-6 MHz	DC-10 MHz*	
Linearity	1% on 1 Vpp	1% on 0.2 Vpp	
Drift	17 $\mu\text{V}/\text{V} \times 10$	\approx 200 ppm/K	
Serial link	RS 485	I2C	
TID	1.7 - 14 krad	100 krad	
LET	10 MeV/mg/cm ²	120 MeV/mg/cm ²	

* Small signal

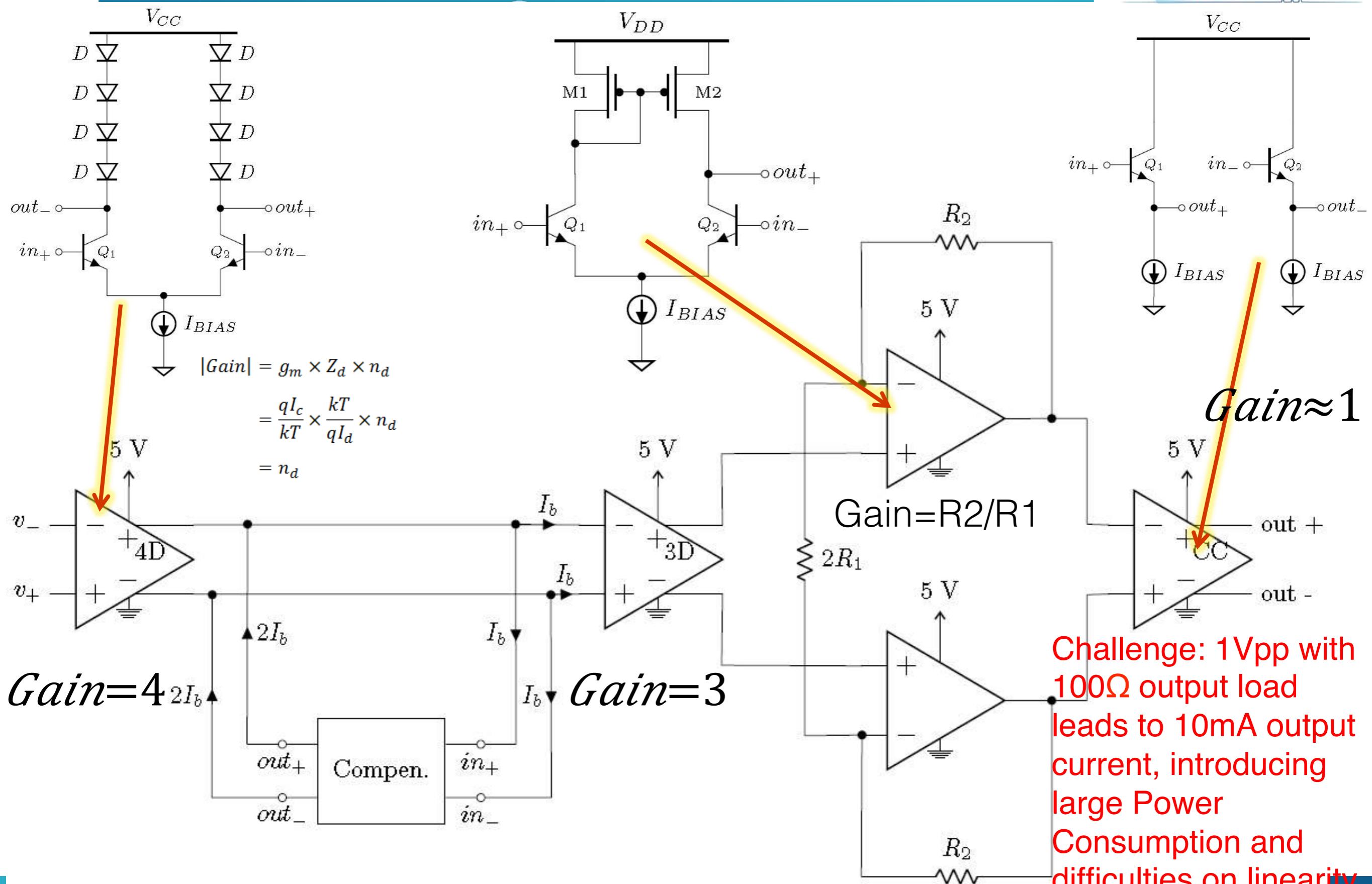


WFEE identified Specifications

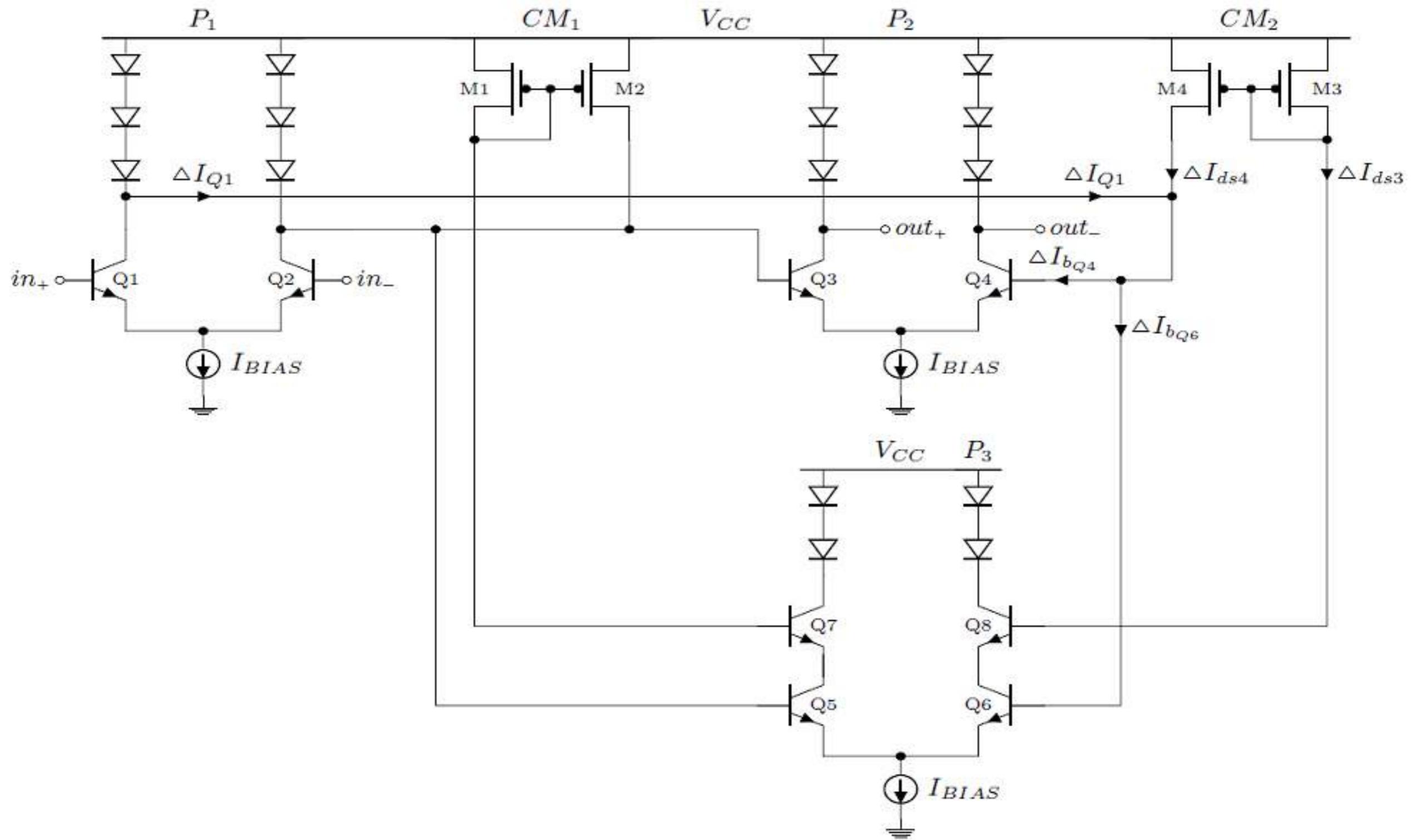
	Identified spec	ASICv1 measure	ASICv2 goal
Gain	80 V/V	3, 8, 24 V/V	80 V/V
Noise	1 nV/ $\sqrt{\text{Hz}}$	\approx 1 nV/ $\sqrt{\text{Hz}}$	< 1 nV/ $\sqrt{\text{Hz}}$
BW	1-6 MHz	DC-10 MHz*	DC-6 MHz**
Linearity	1% on 1 Vpp	1% on 0.2 Vpp	1% on 1 Vpp
Drift	17 $\mu\text{V}/\text{V} \times 10$	\approx 200 ppm/K	< 300 ppm/K
Serial link	RS 485	I2C	I2C _{in} /RS485 _{out}
TID	1.7 - 14 krad	100 krad	-
LET	10 MeV/mg/cm ²	120 MeV/mg/cm ²	-

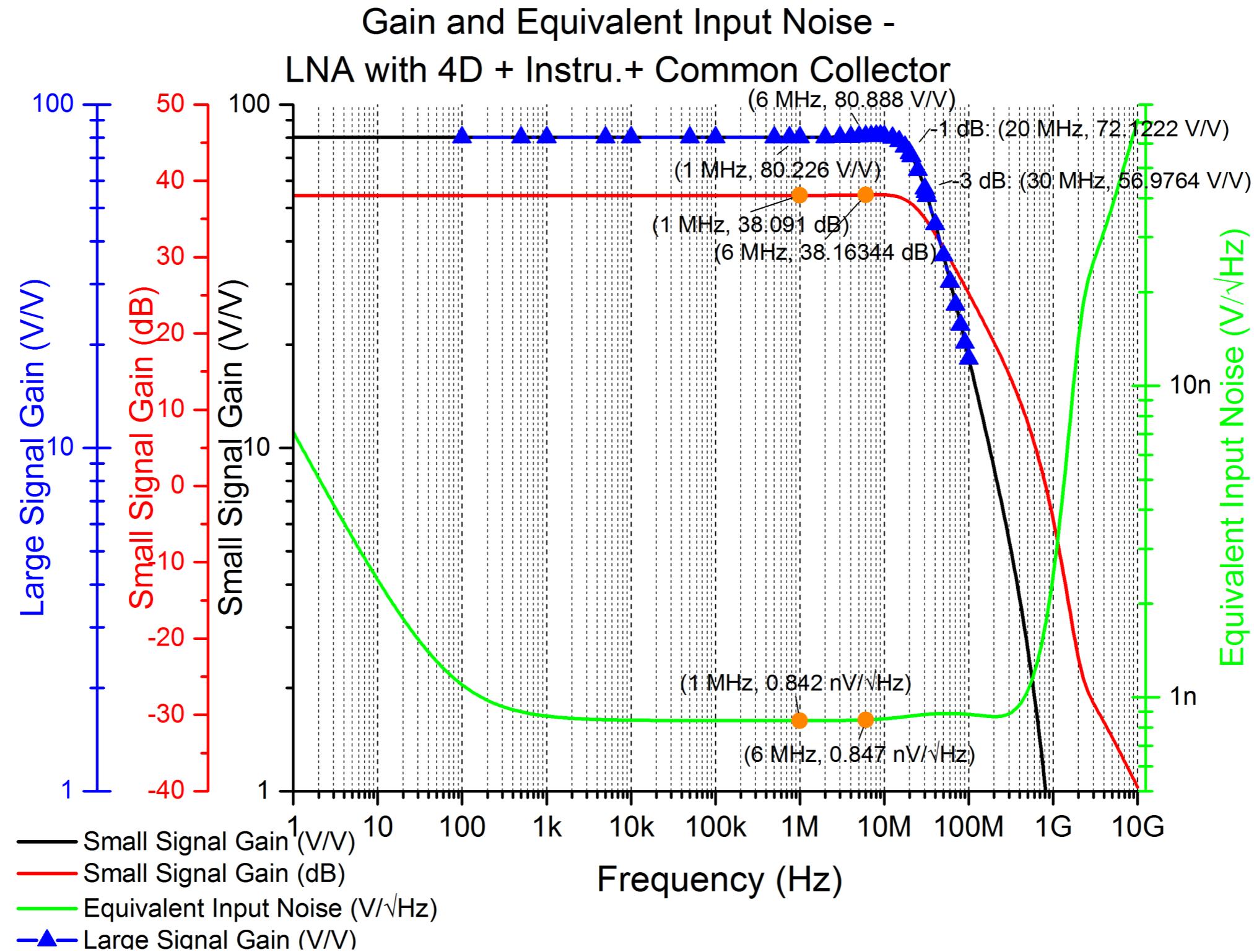
* Small signal

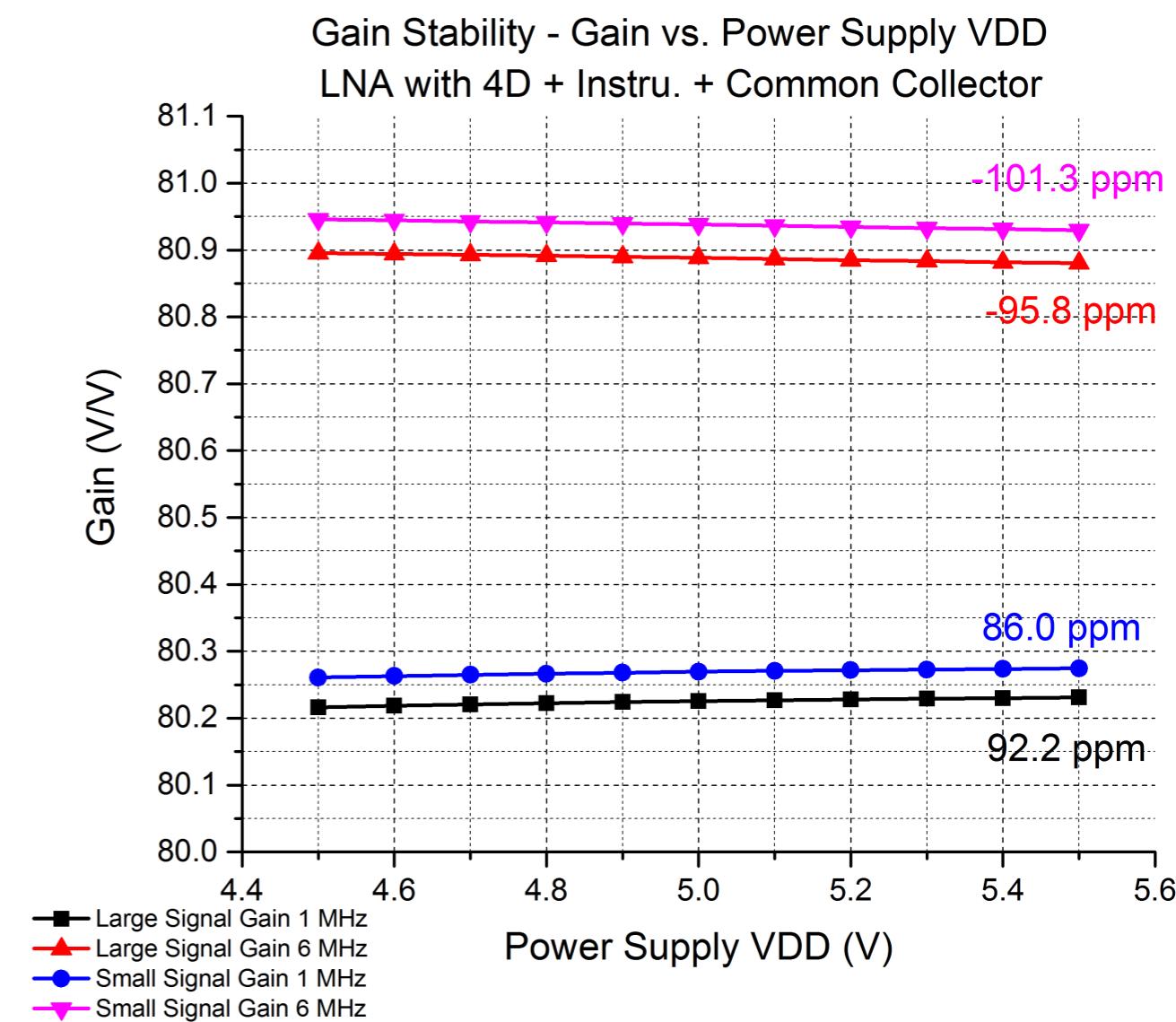
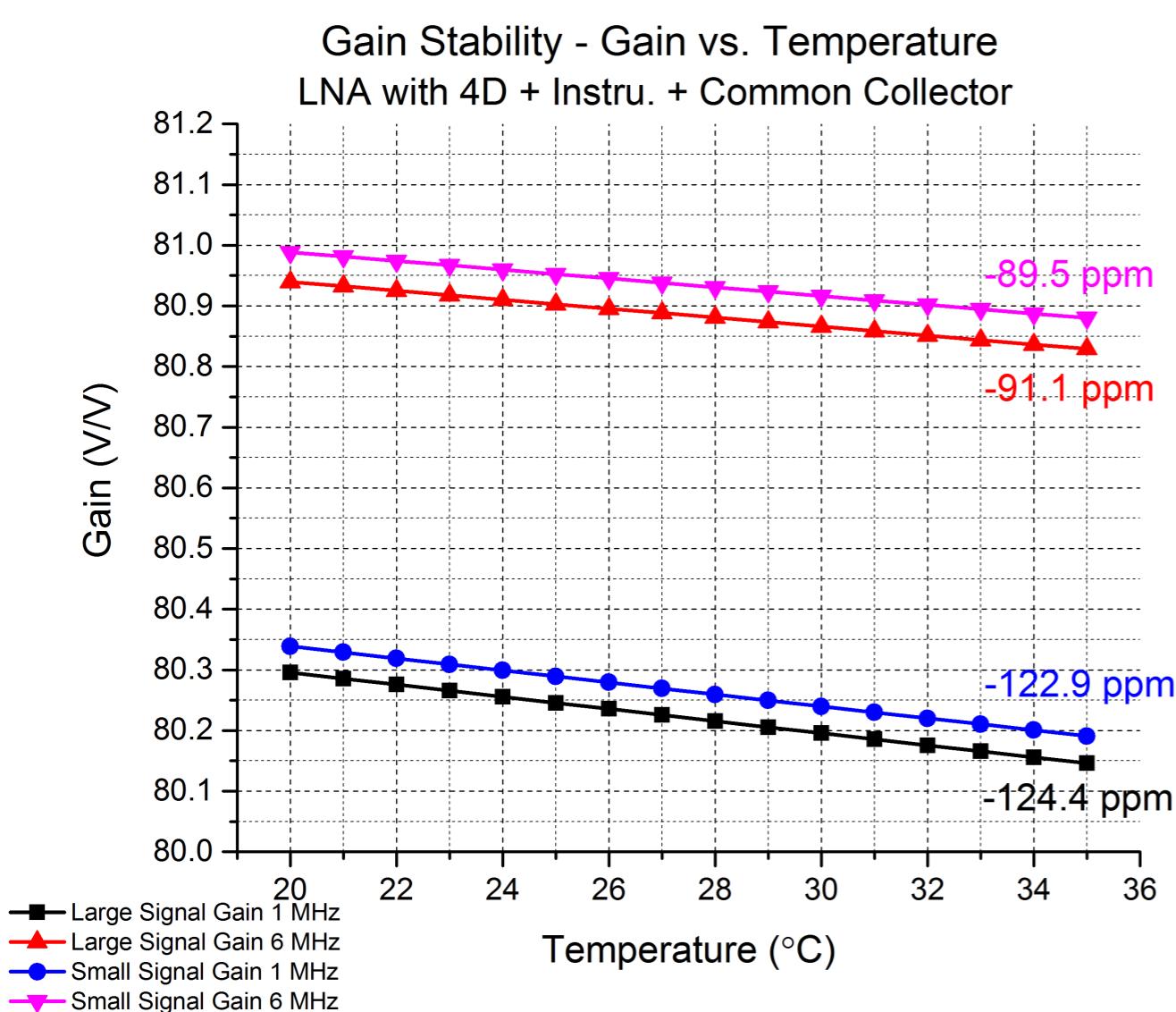
** Small and full dynamic (slew rate)

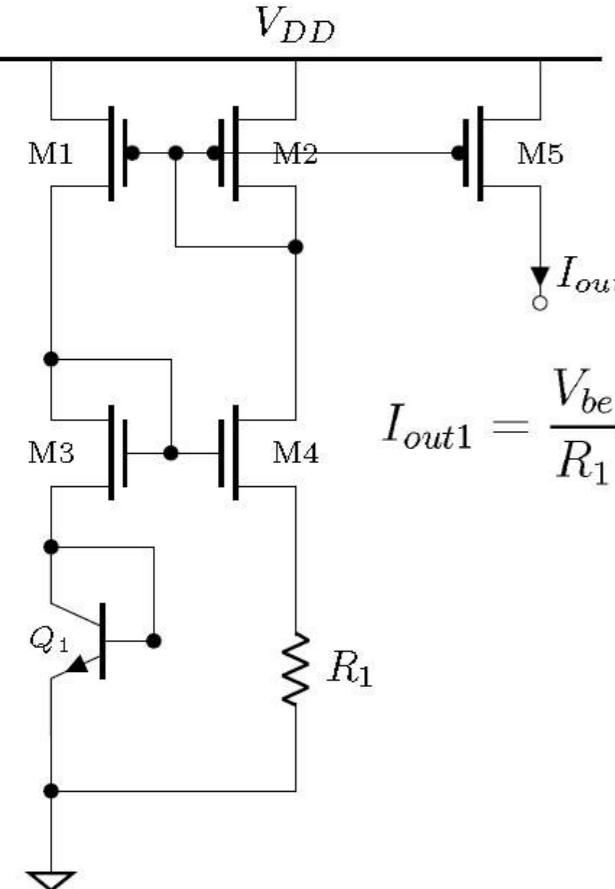


Compensation system of Base current of npn bipolar transistor

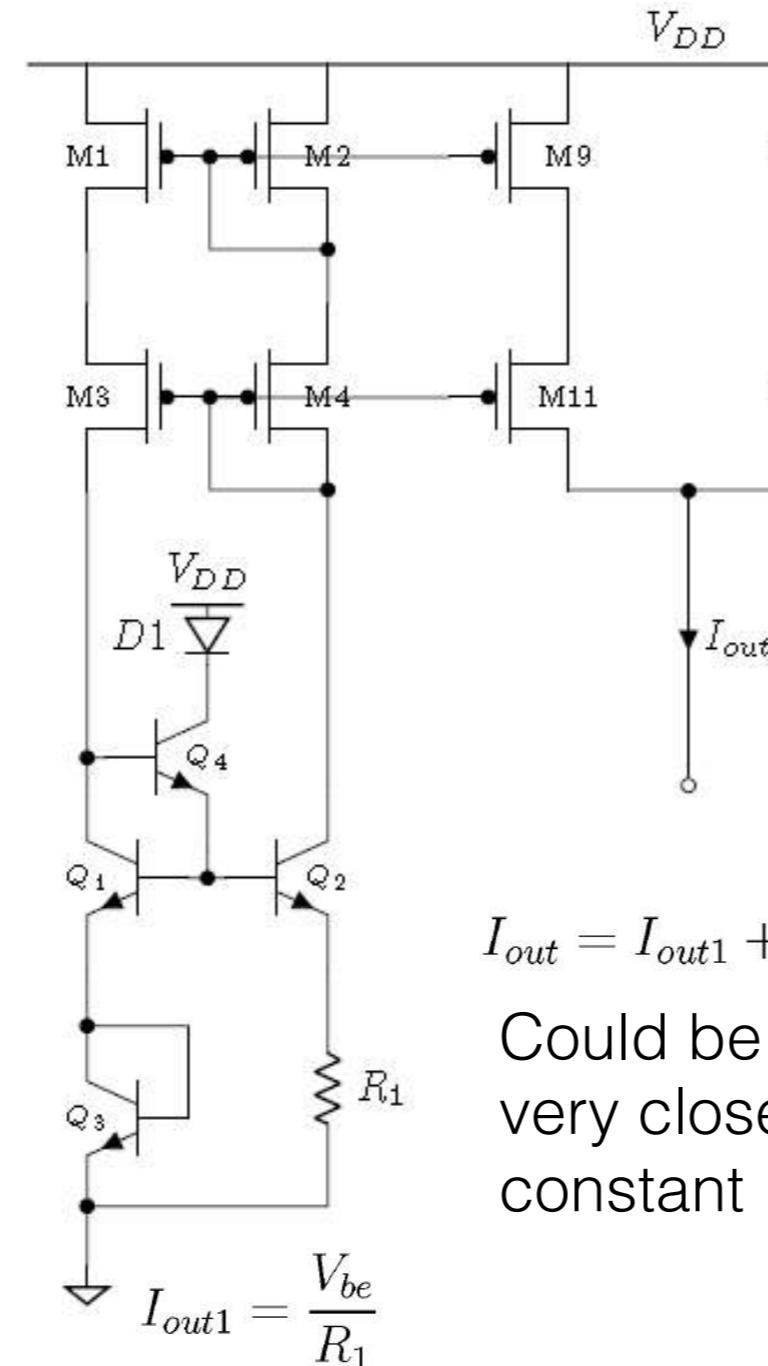






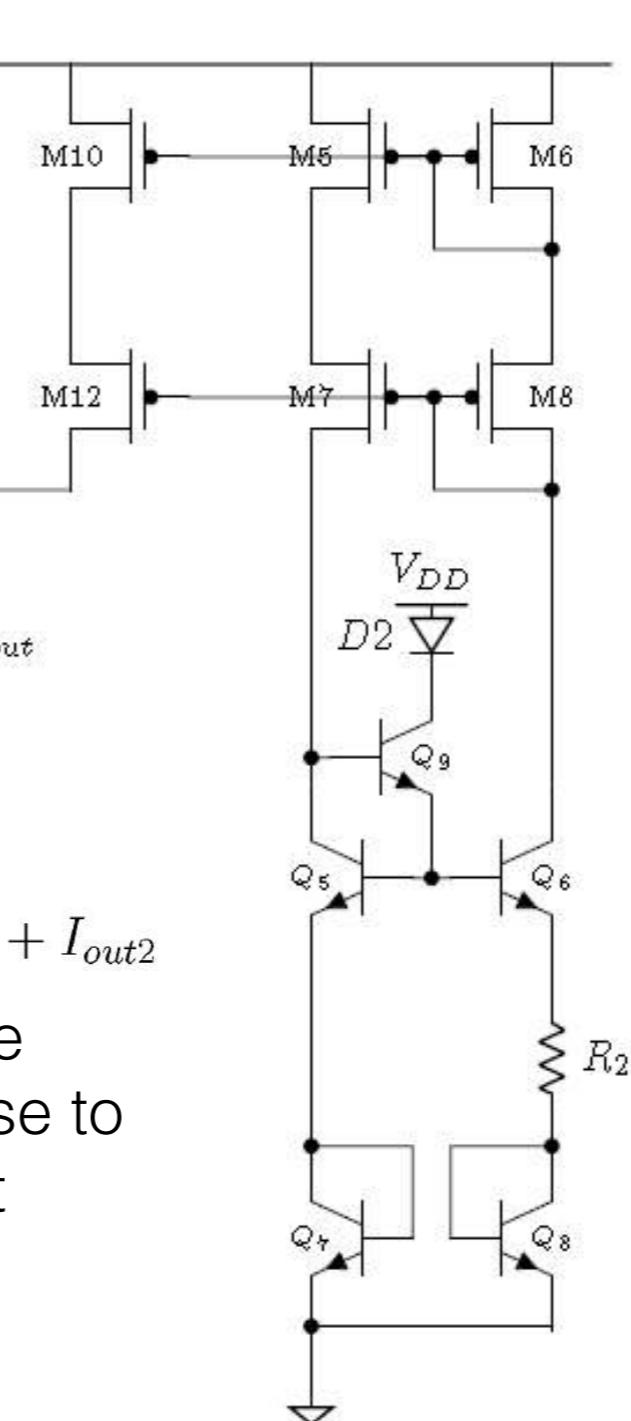


Negative
temperature
coefficient



Negative
temperature
coefficient

$I_{out} = I_{out1} + I_{out2}$
Could be
very close to
constant



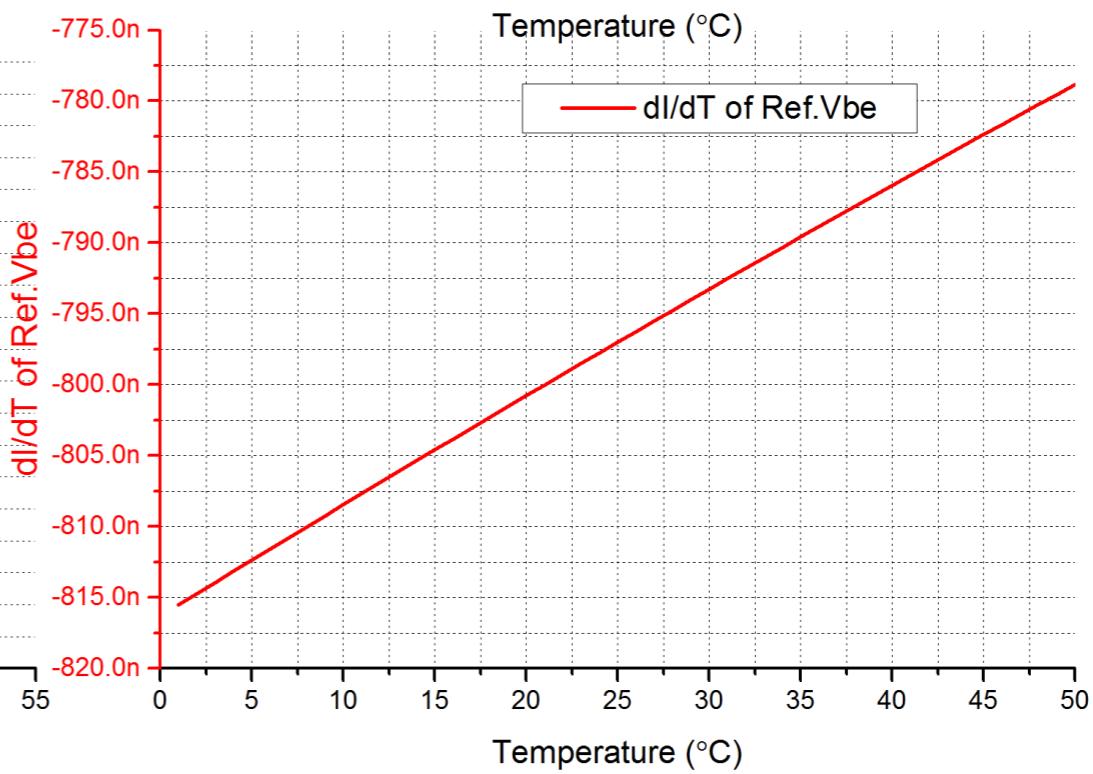
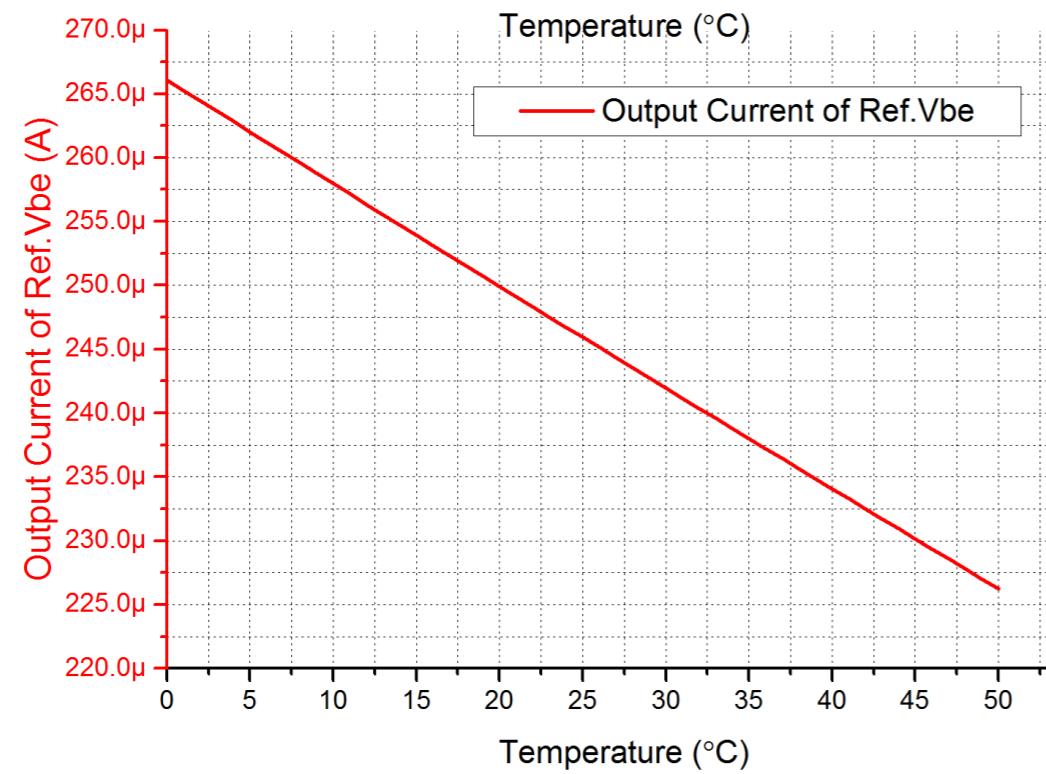
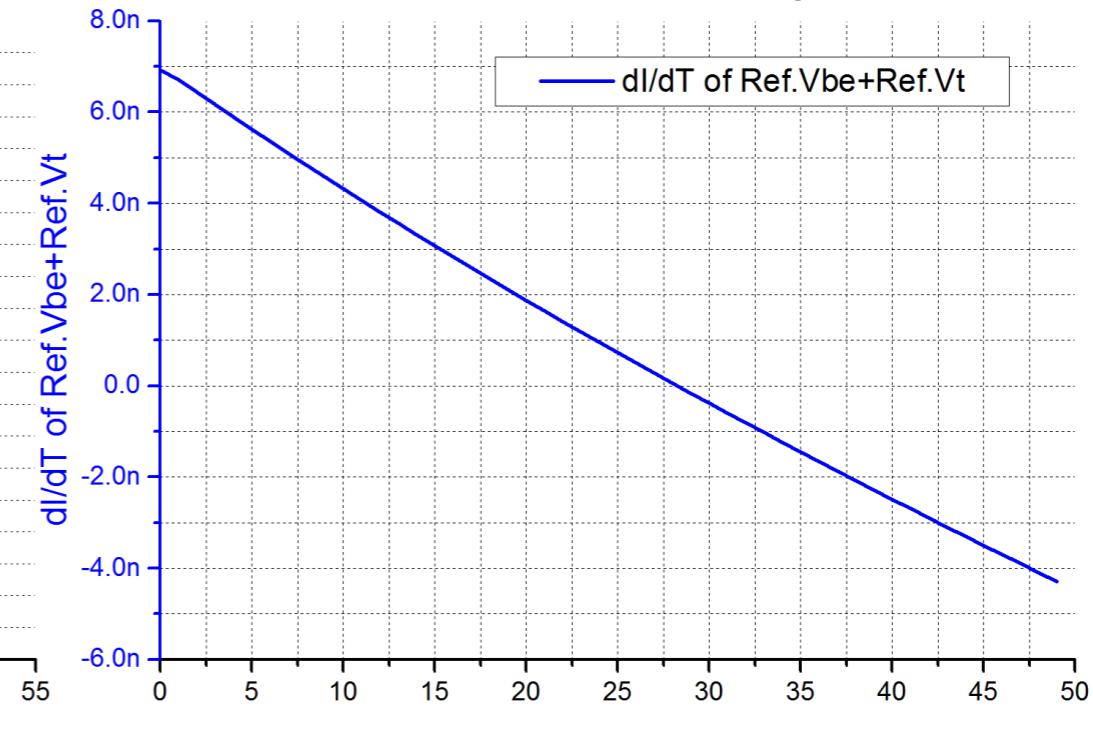
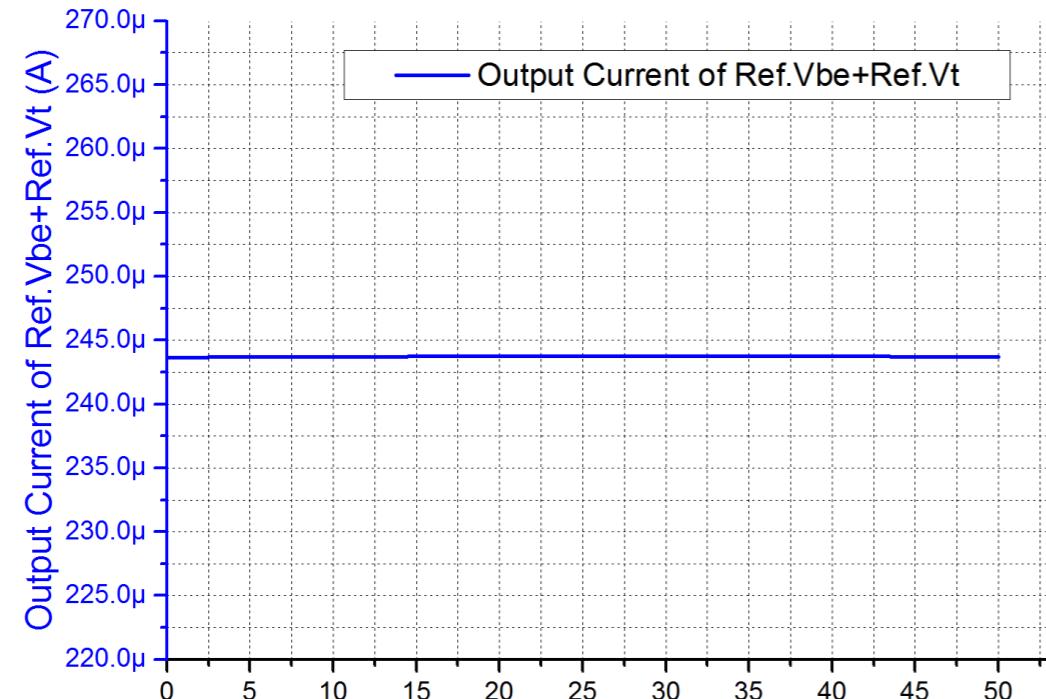
Positive
temperature
coefficient

$$I_{out2} = \frac{\Delta V_{be}}{R_2} = \frac{V_{be7} - V_{be8}}{R_2}$$

$$= \frac{V_T \ln \frac{I_0}{I_s} - V_T \ln \frac{I_0}{mI_s}}{R_2}$$

$$= \frac{V_T \ln m}{R_2}$$

Output Current vs. Temperature for Current Source with Ref.Vt+Ref.Vbe and Current Source only with Ref.Vbe



Simulations, Layout, Post-simulation (before 03/2018)

Measurements (07/2018 -)

	Identified spec	ASICv1 measure	ASICv2 goal
Gain	80 V/V	3, 8, 24 V/V	80 V/V
Noise	$1 \text{ nV}/\sqrt{\text{Hz}}$	$\approx 1 \text{ nV}/\sqrt{\text{Hz}}$	$< 1 \text{ nV}/\sqrt{\text{Hz}}$
BW	1-6 MHz	DC-10 MHz*	DC-6 MHz**
Linearity	1% on 1 Vpp	1% on 0.2 Vpp	1% on 1 Vpp
Drift	$17 \mu\text{V}/\text{V} \times 10$	$\approx 200 \text{ ppm}/\text{K}$	$< 300 \text{ ppm}/\text{K}$
Serial link	RS 485	I2C	$\text{I2C}_{in}/\text{RS485}_{out}$
TID	1.7 - 14 krad	100 krad	-
LET	$10 \text{ MeV}/\text{mg}/\text{cm}^2$	$120 \text{ MeV}/\text{mg}/\text{cm}^2$	-

* Small signal

** Small and full dynamic (slew rate)



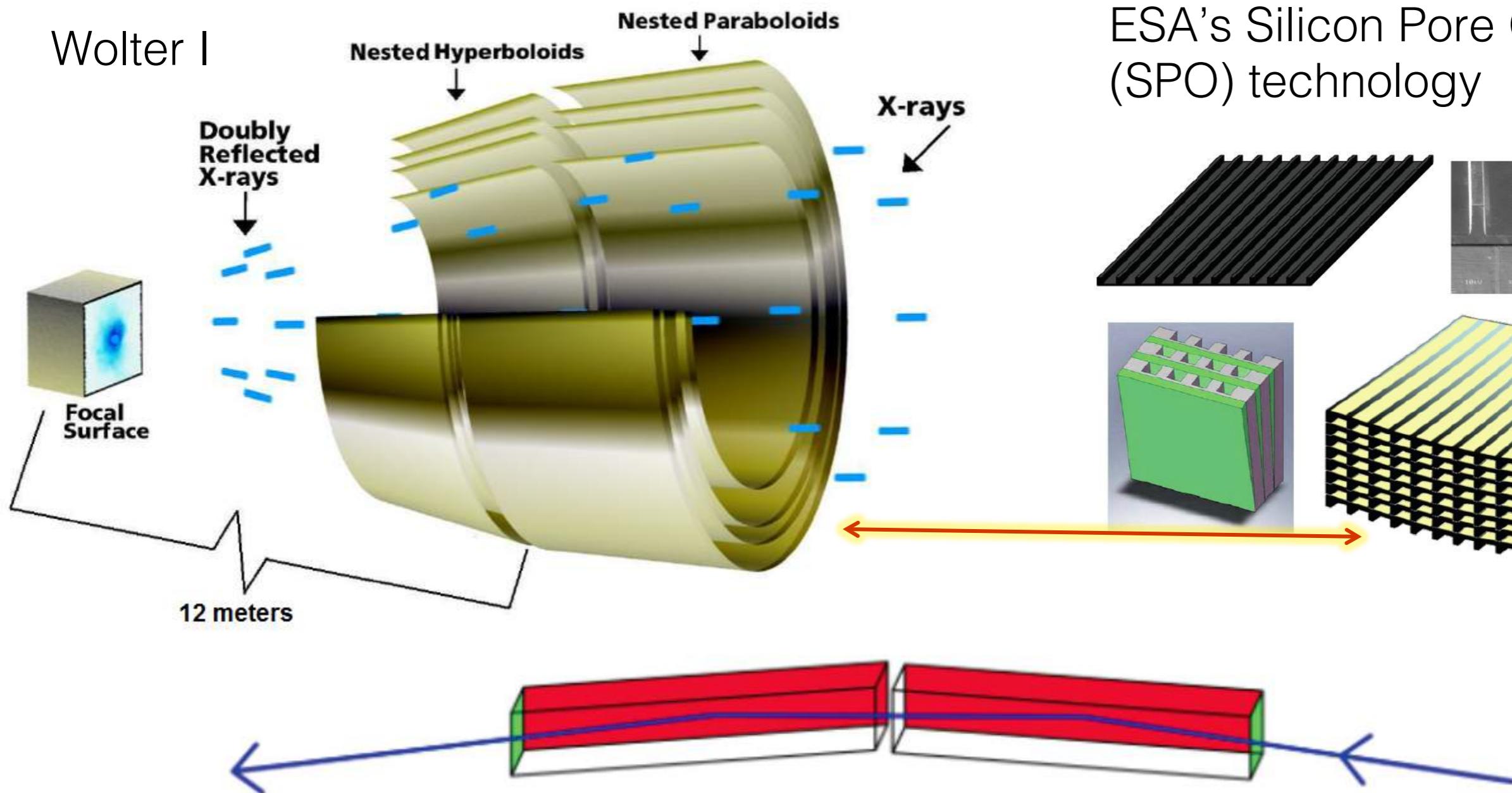
Key parameters and requirements of the Athena+ mission

Parameter	Requirements	Enabling technology/comments
Effective Area	2 m ² @ 1 keV (goal 2.5 m ²) 0.25 m ² @ 6 keV (goal 0.3 m ²)	Silicon Pore Optics developed by ESA. Single telescope: 3 m outer diameter, 12 m fixed focal length.
Angular Resolution	5" (goal 3") on-axis 10" at 25' radius	<i>Detailed analysis of error budget confirms that a performance of 5" HEW is feasible.</i>
Energy Range	0.3-12 keV	Grazing incidence optics & detectors.
Instrument Field of View	<i>Wide-Field Imager: (WFI): 40' (goal 50')</i>	Large area DEPFET Active Pixel Sensors.
	<i>X-ray Integral Field Unit: (X-IFU): 5' (goal 7')</i>	Large array of multiplexed Transition Edge Sensors (TES) with 250 micron pixels.
Spectral Resolution	WFI: <150 eV @ 6 keV	Large area DEPFET Active Pixel Sensors.
	X-IFU: 2.5 eV @ 6 keV (goal 1.5 eV @ 1 keV)	<i>Inner array (10"x10") optimized for goal resolution at low energy (50 micron pixels).</i>
Count Rate Capability	> 1 Crab ¹ (WFI)	<i>Central chip for high count rates without pile-up and with micro-second time resolution.</i>
	1 mCrab, point source (X-IFU) with 90% of high-resolution events	<i>Filters and beam diffuser enable higher count rate capability with reduced spectral resolution.</i>
Target of Opportunity Response	4 hours (goal 2 hours) for 50% of time	<i>Slew times <2 hours feasible; total response time dependent on ground system issues.</i>

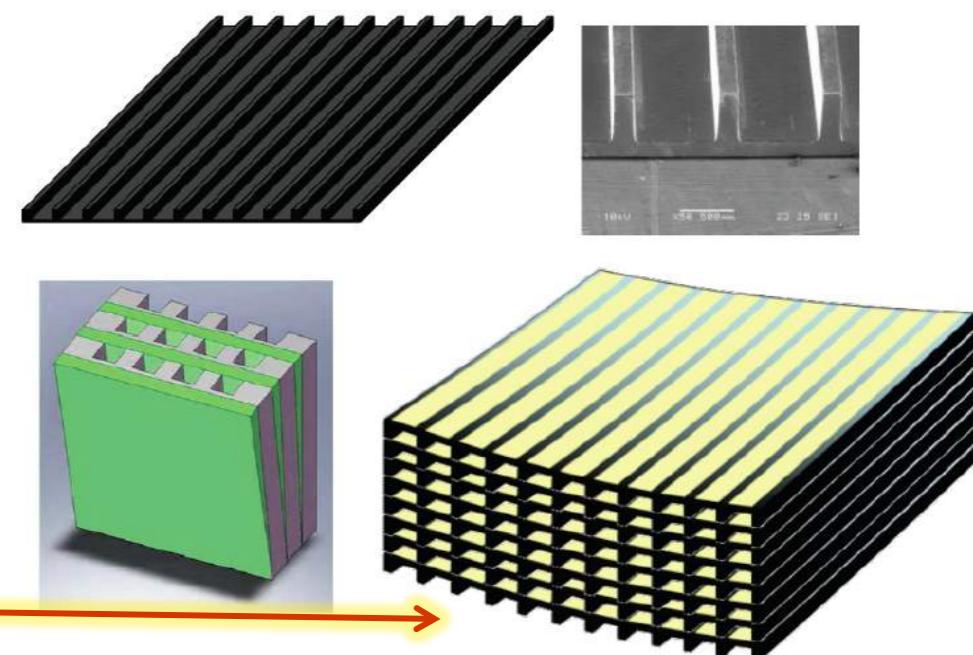
- 3 key components:

- 1) X-Ray telescope - 12m focal length, an effective area about 2 m^2 at 1 keV

Wolter I

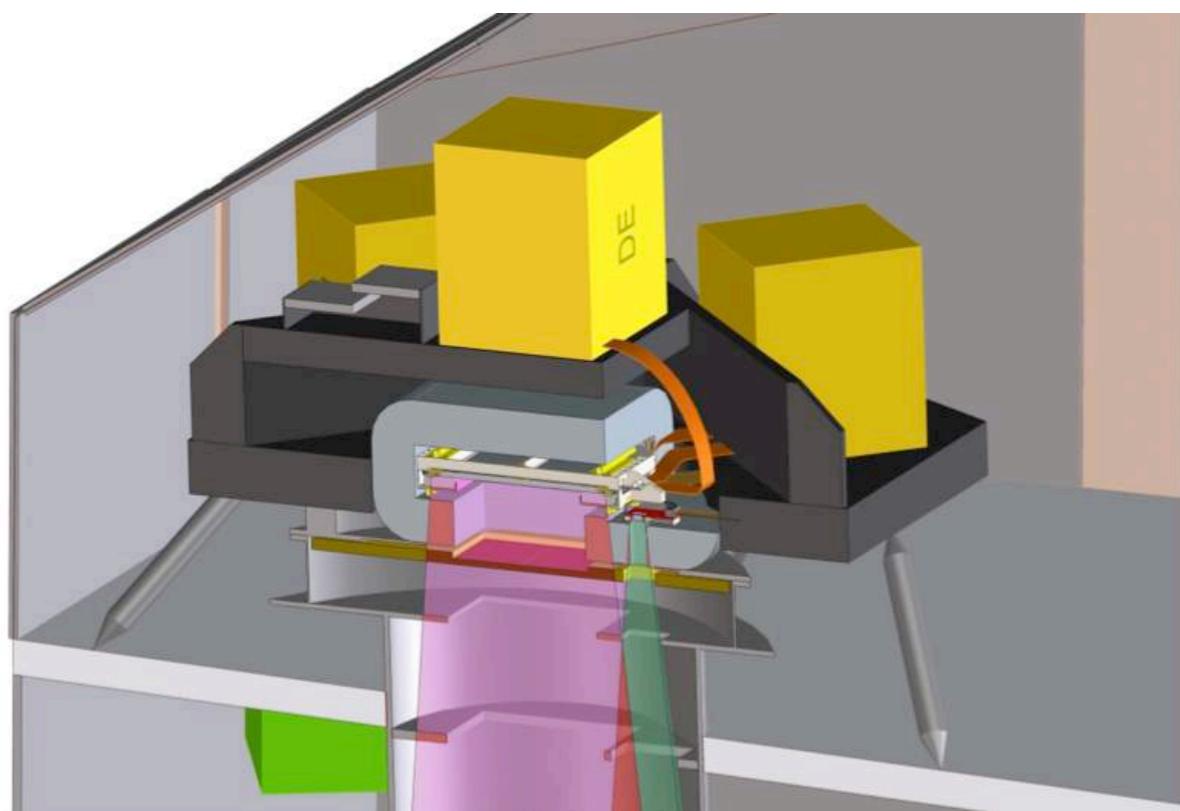
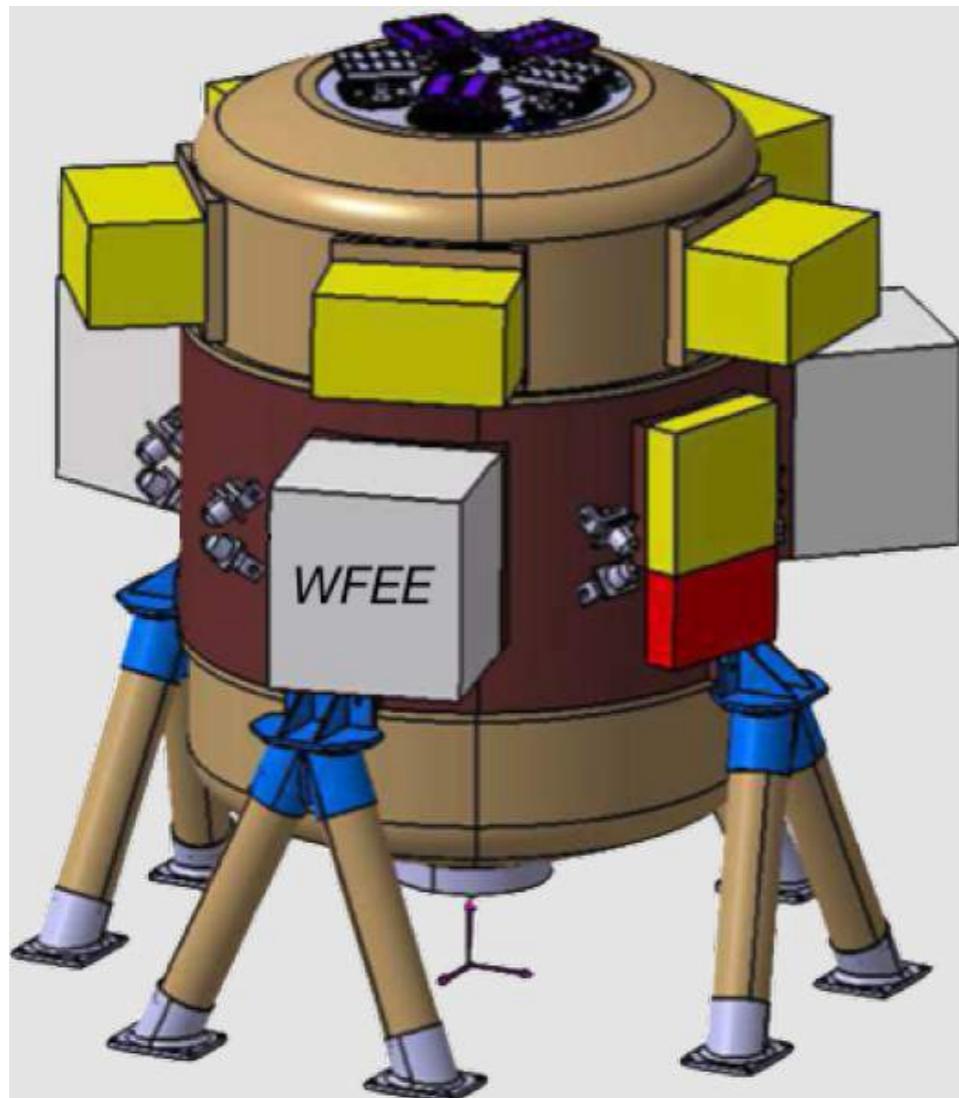


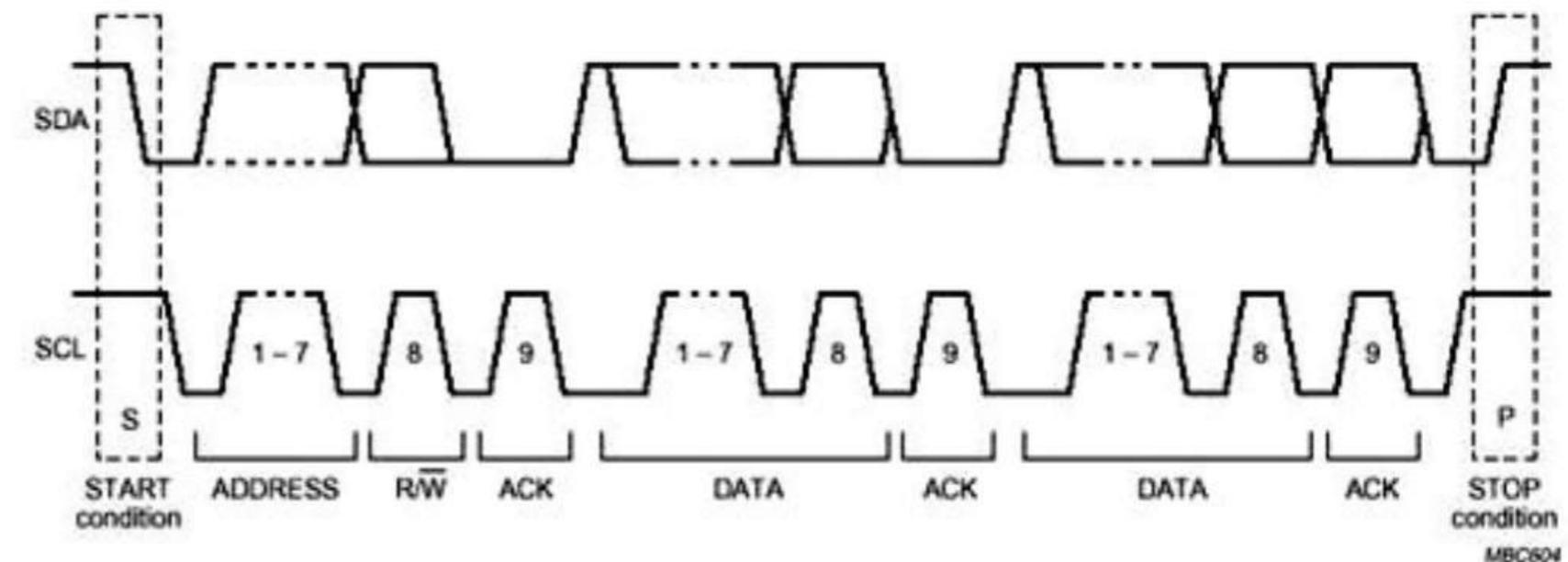
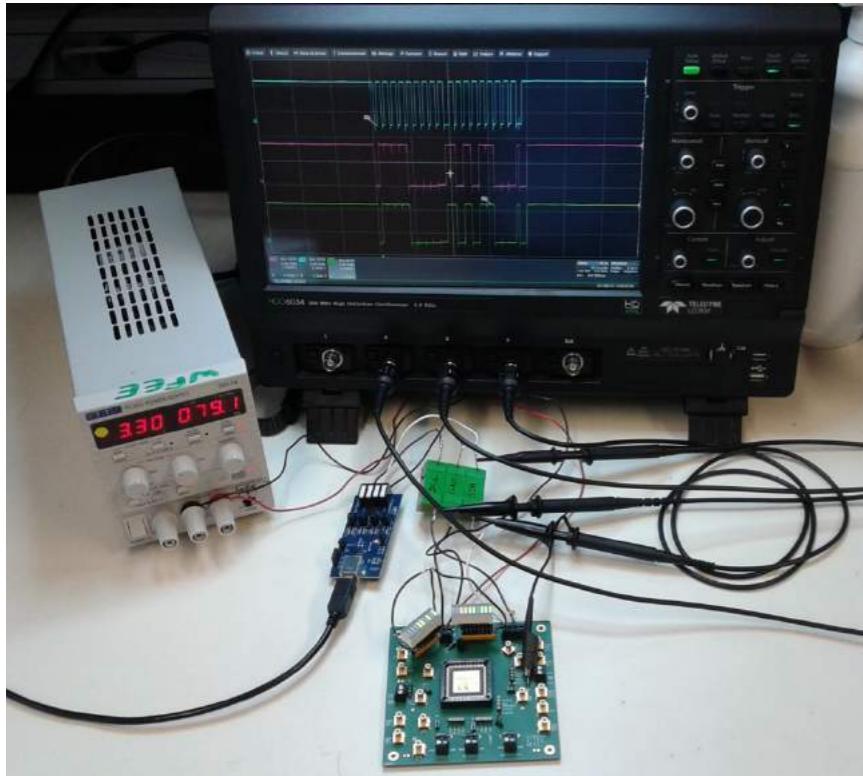
ESA's Silicon Pore Optics (SPO) technology



- 2)X-IFU (X-ray Integral Field Unit) : Transition Edge Sensors (TES), 2.5 eV high spectral resolution: $E < 7 \text{ keV}$
- 5"pixels, field of view 5 arc minutes

- 3)WFI (Wide Field Imager) for high count rate, large field of view, 0.2-15 keV energy band
- Silicon-based detector using DEPFET Active Pixel Sensor (APS) technology





I2C Decoder Test - Short SDA and SCL lines (DC Coupling) - with Pull-up Resistors 1kΩ - SCL 100kHz - Data 0xAC

